BALASHOV, M.A.; VORONKOV, B.S.; YELAGIN, Ye.B.; KISELEV, L.N.; KOLOSOV, S.P.; LEONT'YEVA, V.P.; NEFEDOVA, V.I.; STROMILOV, V.M.; SOKOLOV, N.I.; TISHCHENKO, N.M.; UDALOV, N.P.; PETROV, B.N., akademik, red.; CRICORASH, K.I., red. izd-va; ROZHIN, V.P., tekhn. red.

[Handbook on the design of components and systems of automatic control; a manual for the preparation of course and diploma projects] Rukovodstvo po proektirovaniiu elementov i sistem avtomatiki; posobie po kursovomu i diplomnomu proektirovaniiu [By] M.A.Balashov i dr. Pod red. B.N.Petrova. Moskva, Gos. nauchno-tekhn. izd-vo Oborongiz. No.4. 1961. 311 p. (MIRA 15:3)

1. Moscow. Aviatsionnyy institut imeni Sergo Ordzhonikidze. (Automatic control) (Electronics)

S/194/62/000/002/059/096 D273/D301

9,2540

AUTHOR:

Kiselev, L. N.

TITLE:

Computing converters of potential based on transistors

PERIODICAL:

Referativnyy zhurnal, Avtomatika i radioelektronika, no. 2, 1962, abstract 2-5-51a (V sb. Poluprovodnik, pribory i ikh primeneniye, no. 7, M., "Sov. radio",

1961, 275-288)

TEXT: Analysis of two-emitter semiconducting converters has been carried out for two cases, when the permeability of the core does not depend on the magnetizing current (core with air gap) and when the transistor core has an ideal square magnetization curve (to-roidal cores). Most applications use converters based on toroidal permalloy cores. Computation of the working rate of such converters is made from the given voltage supply, the voltage and power outputs, and the frequency of the converter. The working rate of the converters and the elements of the inverter circuit are determined; transistor calculations are carried out. The computation method

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S/194/62/000/002/059/096
Computing converters of ...

D273/D301

gives good agreement with practical tests for converters working in the 5 - 500 watt range. 5 references. / Abstracter's note: Complete translation. /

KISELEV, lev Nikolayevich; MIKITIN, Viktor Borisovich

Frequency control of a square-wave voltage self-oscillator. Izv.
vys.ucheb.zav.; elektromekh. 7 no.10:1225-1230 164.

l. Starshiye inzh. Moskovskogo aviatsionnogo instituta.

(MIRA 18:1)

S/194/62/000/004/014/105 D222/D309

9,2530

AUTHOR:

Kiselev, L. N.

TITLE:

Transistor-magnetic average-current pulse amplifier

PERIODICAL:

Referativnyy zhurnal, Avtomatika i radioelektronika, no. 4, 1962, abstract 4-2-10y (Poluprovodnik. pribory i ikh primeneniye, no. 7, M., Sov. radio, 1961, 207-228)

TEXT: Some variants of average-current pulse amplifier circuits supplied from a DC voltage source are considered. A classification of such circuits is given. A comparative analysis of the following amplifier circuits is carried out: Pulse amplifiers with ohmicinductive loading, magnetic amplification with pulse duration modulation. The circuit of a controllable oscillator with current transformer and differential loading, and that of a magnetic amplifier supplied by a DC voltage, are given. It is shown that these pulse amplifier circuits for average-current amplification have a greater efficiency, good linearity and considerable lifetime com-

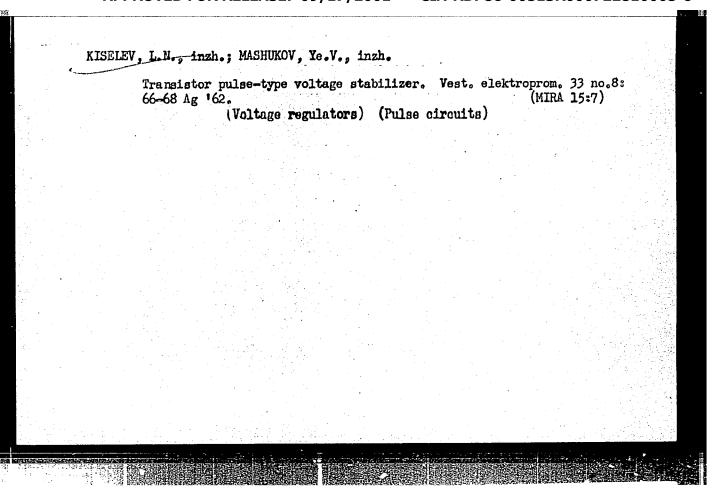
Card 1/2

Transistor-magnetic ...

S/194/62/000/004/014/105 D222/D309

pared with other types. Computational formulas are given for transistors working in the switching regime. The basic relationships for the design of the given circuits are derived. The choice of a suitable circuit depends on the actual amplification requirements. 11 figures. 6 references. / Abstracter's note: Complete translation. 7

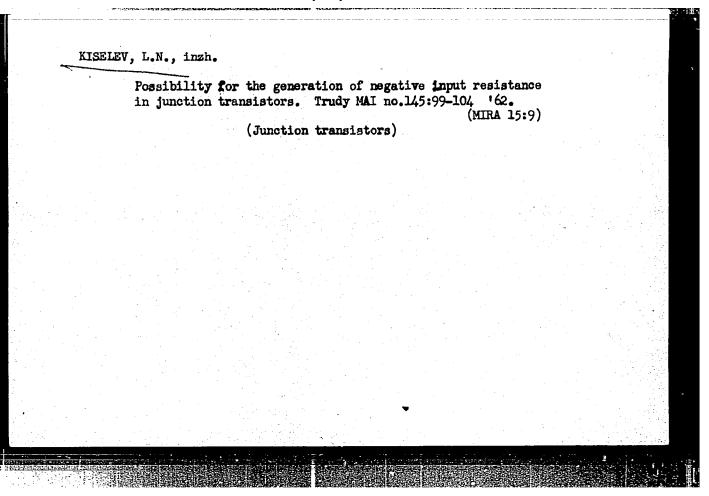
Card 2/2

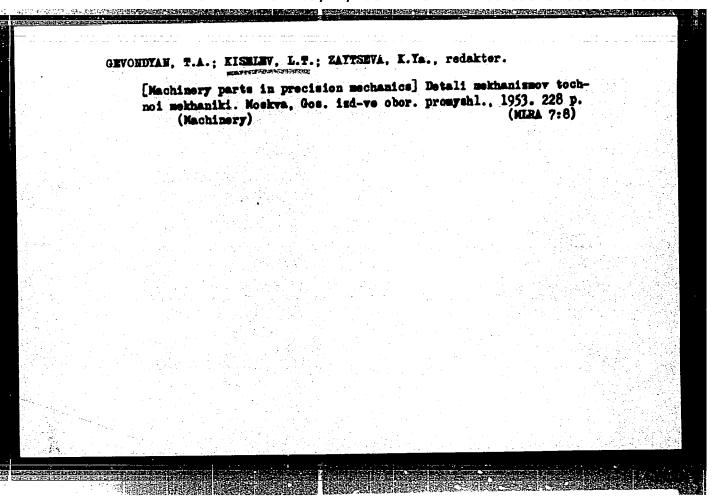


YEVDOKIMOV, V.G.; FETTGIN, V.I.; PYZHOV, V.S.; prinimali uchastiye: SMIRNOV, V.M.; KISELEV, L.W.; SHUNILOV, A.S.; VINOKUROV, V.K.; TIKHONOV, N.A.

Investigating granulators as controlled systems. TSvet. met. 35 no.6:
41-46 Js **162.**

(Ore dressing) (Granular materials)

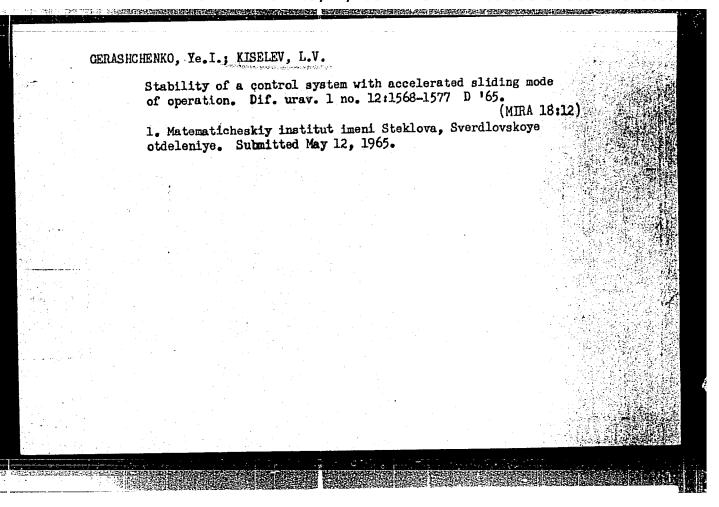


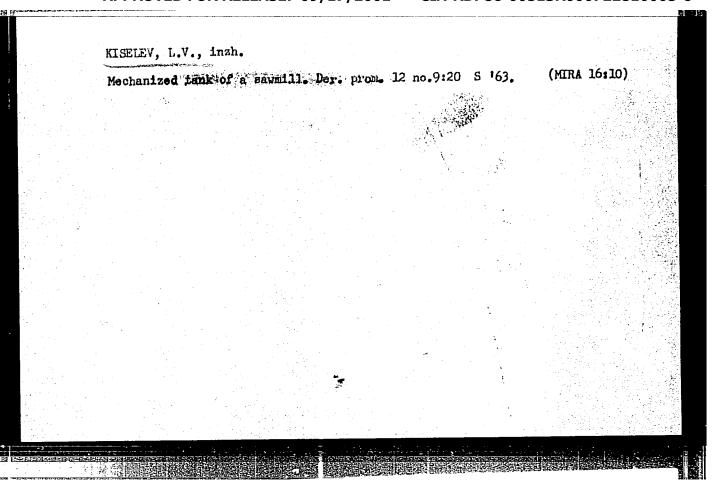


GEVONDYAN, Tigran Arutyunovich; KISELEV, Lev Timofeyevich; RYABOV, B.A., doktor tekhm. nauk, prof., retsenzent; ZAKAZNOV, N.P., kand. tekhm. nauk, retsenzent; DOEROGURSKIY, S.O., doktor tekhm. nauk, prof., zasl. deystel' nauki i tekhniki, red.; YELISEYEV, M.S., red. izd-va; MODEL', B.I., tekhm. red.

[Devices for measuring and recording vibrations] Pribory dlia immerentia i registratsii kolebanii. Moskva, Mashgiz, 1962.

467 p. (Vibration-Measurement)





sov/96-59-6-2/22

Deych, M.Ye., (Dr. Tech.Sci.), Kazintsev, F.V., Abramov, V.I., Kiselev, L.Ye. and Filippova, V.G. AUTHORS:

(Engineers)

An Investigation of Turbine Stages with Long Blades of TITLE:

Constant Profile under Variable Conditions (Issledovaniye

peremennogo rezhima turbinnykh stupenei s dlinnymi

lopatkami postoyannogo profilya)

PERIODICAL: Teploenergetika, 1959, Nr 6, pp 8-17 (USSR)

ABSTRACT: This article describes the results of tests on four single-row stages with relatively long blades of constant profile, fitted to an experimental turbine. efficiency of single-row stages depends on a number of geometrical and operating conditions: the configuration, pitch and angles of installation of the blades, the ratio of the flow areas, the velocity ratio and the Mach and Reynolds numbers. The tests described here were made to study the influence of these factors on the efficiency. The stages had a d/2 ratio + 7.73 which is the limiting value for cylindrical blading. The four stages investi-

gated employed two types of guide vanes (TS-1A and TS-2A) and two types of working blades (TR-2A and TR-3A). Card 1/8

APPROVED FOR RELEASE: 09/17/2001 CIA-RDP86-00513R000722810005-6"

sov/96-59-6-2/22

An Investigation of Turbine Stages with Long Blades of Constant Profile under Variable Conditions

The principal geometrical characteristics of the blading are given in Table 1. All the stages used welded diaphragms of 400 mm mean diameter with guide vanes 48 mm high and working blades 51.7 mm high. The measuring equipment used is briefly described. The stages were tested with ratios of back pressure to inlet pressure of 0.9 to 0.54, which corresponds to a Mach number range of 0.4 to 1.0. The tests were made with constant back pressure. The influence of diaphragm leakage on the efficiency and the degree of reaction at root and tip sections were investigated. The quantity of leakage steam ranged from 0.8 to 3.5% of the flow through the guide vanes. The influence of the Reynolds number on the stage characteristics was investigated in three of the stages, with Reynolds numbers ranging from 3 x 105 to 7 x 105. The maximum error in determining the stage efficiency was between 0.4 and 0.6%. The influence of compressibility on the stage efficiency and degree of reaction is then considered. Stage efficiency graphs as

functions of velocity and pressure ratios are given in

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An Investigation of Turbine Stages with Long Blades of Constant Profile under Variable Conditions

Fig 1: it will be seen that for each stage there is a pressure ratio that gives maximum efficiency. Values of the best pressure ratio, the highest efficiency, and the change in efficiency as the pressure ratio deviates from the optimum value, are tabulated in Table 2. The curves in Fig 1 show that the efficiency is fairly stable as the velocity ratio changes, indicating that stages with guide vanes type TS-2A have a flatter characteristic as a function of the velocity ratio. This is because the ratio of the blade area to the guide-vane area is lower and there is consequently more reaction in stages with these guide vanes. Curves of stage efficiency as a function of Mo with constant velocity ratio are given in Fig 2a, and curves of efficiency as function of the available heat drop with the speed constant in Fig 2b. From consideration of these curves it is concluded that the stage efficiency is reasonably stable. the pressure distribution over the pitch of the guide Card 3/8 vanes at the tip and root sections respectively are given in Figs 3a and 3b. Corresponding curves under static

sov/96-59-6-2/22 An Investigation of Turbine Stages with Long Blades of Constant

Profile under Variable Conditions conditions and in the presence of a working wheel are given in Fig 3c. It will be seen that the static pressure field is very irregular. Graphs of the reaction at root and tip sections as a function of the velocity ratio are given in Fig 4. It will be seen that in most cases the reaction is negative at the blade roots. These tests were made in the absence of diaphragm leakage. presence of negative reaction at the blade roots has no appreciable influence on the stage efficiency. curves of distribution of reaction over the radius for stages KD-2-2A and KD-2-3A at various values of velocity ratio and constant pressure ratio are given in Fig 5. The curves were constructed from experimental values of the loss factors at different sections of the guide vanes and reaction in the root section, using formula (2). It will be seen that the agreement between the

experimental and calculated values of reaction is satisfactory. Graphs of the relative difference of root and tip reaction as a function of the relative change in the velocity ratio are given in Fig 6. Over the range Card 4/8

SOV/96-59-6-2/22

An Investigation of Turbine Stages with Long Blades of Constant Profile under Variable Conditions

of change of velocity ratio from -0.2 to +0.2 this relationship is given by formula (3). It was found that there is a certain range of Reynolds and Mach numbers and of diaphragm leakage for which formula (3) remains valid, as will be seen from the results plotted in Fig 6. Formula (3) can serve as a basis for two methods of designing stages with long blades operating under variable conditions, as is briefly explained. The influence of Reynolds number on the stage efficiency is then considered. A series of tests was made on the three stages. The influence of the Reynolds number was thereby evaluated in stages having different degrees of reaction at the root and middle sections. The test results, plotted in Fig 7, are discussed at some length. found that the influence of the Reynolds number is Graphs of the greatest when the velocity ratio is high. relationship between the maximum stage efficiency and the Reynolds number appear in Fig 8, and graphs showing the influence of the Reynolds number on the reaction at the root and tip sections of the three stages are plotted in

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SOV/96-59-6-2/22

An Investigation of Turbine Stages with Long Blades of Constant Profile under Variable Conditions

Fig 9. Graphs of the flow coefficients as a function of Reynolds number are plotted in Fig 10. The influence of diaphragm and leakage is then considered. In order to determine the influence of diaphragm leakage on the stage characteristics, steam was delivered from the steam chest to the space between the disc and diaphragm in amounts up to 5% of the main flow. Graphs of the changes in efficiency as functions of leakage are plotted in Fig 11. Graphs of tip and root reaction, and flow coefficient as function of velocity ratio and a graph of the influence of leakage on the change in stage reaction, are plotted in Figs 12a and 12b respectively. It is found that increase in Reynolds number and decrease in leakage The results of a reduces both root and tip reaction. detailed study of the flow structure in stage KD-2-24 are discussed. The main conclusions are that the ratio of the flow area of the working blades to that of the guide vanes has a considerable influence on stage efficiency. Alterations of the blade root reaction from + 5% to zero had little influence on the stage efficiency. The presence of low negative reaction

Card 6/8

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An Investigation of Turbine Stages with Long Blades of Constant Profile under Variable Conditions

caused some reduction in stage efficiency. With increase in the compressibility (Mach number) the efficiency first rises and then falls. The optimum value of the Mach number depends on the stage geometry and particularly on the sta the area ratio and the type of blades used. As the Mach number increases, so does the reaction. Detailed investigation of the flow structure showed that alteration of the area ratio alters the losses in the working blades and the discharge velocity loss. The flow was found to be very uneven at the outlet section of the guide vanes. It was established that over a certain range of Mach numbers, rotation of the runner has no important influence on the velocity distribution over the pitch of the guide It follows from this that stage calculations based on static steam tests on full-scale diaphragms are

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APPROVED FOR RELEASE: 09/17/2001 CIA-RDP86-00513R000722810005-6"

SOV/96-59-6-2/22

An Investigation of Turbine Stages with Long Blades of Constant Profile under Variable Conditions

reliable provided that the Mach and Reynolds numbers are equal in the actual and model conditions.

There are 12 figures, 2 tables and 5 Soviet references.

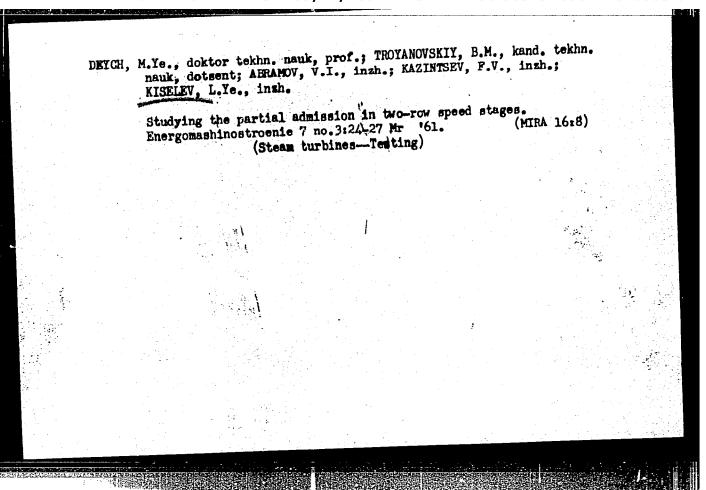
ASSOCIATION: Moskovskiy energeticheskiy institut (Moscow Power Institute)

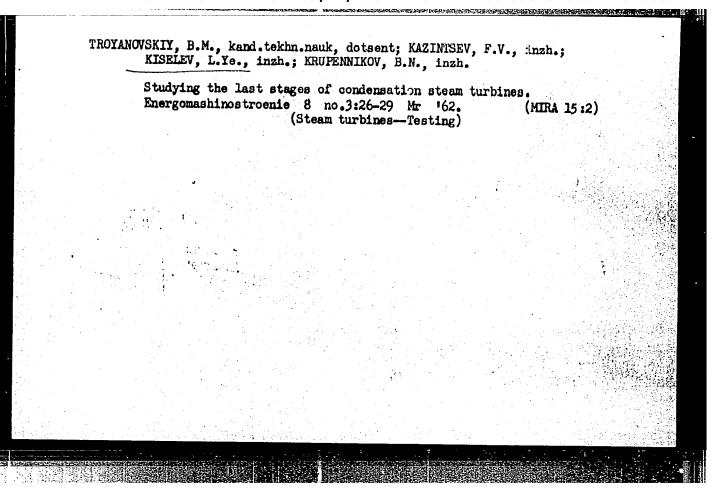
Card 8/8

TROYANOVSKIY, B.M., kand.tekhn.nauk; KISELEV, L.Ye., insh.; FILIPPOVA, V.G., insh.

Methods for calculating two-row velocity stages. Energomashinostroenie 6 no.513-6 My '60.

(Steam turbines)





L 39483-65

ACCESSION NR: AP5011717

UR/0096/64/000/011/0026/0030

AUTHOR: Deych, M. Ye. (Doctor of technical sciences, Professor); Troyanovskiy, 7 B. M. (Candidate of technical sciences); Kiselev, L. Ya. (Engineer); Krupennikov,

B. N. (Engineer)

TITLE: Investigation of an annular turbine grill of large fan shape

SOURCE: Teploenergetika, no. 11, 1964, 26-30

TOPIC TAGS: electric power engineering, power plant component

ABSTRACT: In the Laboratory of Steam and Gas Turtines of the Moscow Power Engineering Institute (MPEI), investigations were made of annular multinozzle grills with d_{cp}/l = 2.5 at various angles of taper of the peripherial meridian line. The tests were conducted in a circular wind tunnel in air at a maximum subsonic speed of M = 0.84. A grill with a variable chord and t = const proved to be highly effective (up to M = 0.86) during the regimes considered. Separation of flow was not observed in any of the grills, which differs from Bammert's conclusions /k. Bammert, H. Klaeukens, Ingenieur-Archiv, Bd XVII, 1949/. This confirms the explanation by the present writers of the separation in certain annular grills, by the flow

Card 1/2

L 39483-65 CCESSION NR: AP5011717		
n the root zone by the he augle of opening of the grill proved to be an iguration with respect to affectiveness of the initialization with distributions of the distributions	the presence in it of diffusion since peripherial line $V_p = 32^{\circ}$, wall, which is explained by the to height ($t = const$, $b = var$) so tial grill TS-IA MPEI. The test buted parameters of the flow vith the most prevalent calculation method of calculation used provided the provided peripherical contents.	optimum grill con- nd the high s showed a large th respect to n with a simpli-
ried Euler formula. The conformity with the expended for graphs.	rimental results. Orig. art. ha	as: 7 formulas, 1 ligato,
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DEYCH, M.Ye., doktor tekhn. nauk, prof.; TROYANOVSKIY, B.M., kand. tekhn. nauk; KISELEV, L.Ye., inzh.; KRUPENNIKOV, B.N., inzh.

Study of an annular large-fan turbine cascade. Teploenergetika li no.11:26-30 N '64. (MIRA 17:12)

1. Moskovskiy energeticheskiy institut.

DEYCH, M.Ye., doktor tekhn.nauk, proffe; KISELLY, L. Tegg. 1850.; KRUFESNIKCY, B.W., inzh.

Effect of the angle of departure on the characteristics of athair turbine lattices with large fan pattern. Izv.vys.ucheb.zav.; energ. 8 no.10:56-62. 0 155. (MTRA 18:10)

1. Moskovskiy ordena Lenina energeticheskis Institut. Poedataviens kafedroy parovykh i gazovykh burbing.

L 18564-66 EWT(m)/EWP(w)/EWP(v)/T-2/EWP(k)/ETC(m)-6 ACC NR: AP6006428 SOURCE CODE: UR/0143/65/000/010/0056/0062 AUTHOR: Deych, M. Ye. (Doctor of technical sciences, Professor); Kiselev. (Engineer); Krupennikov, B. N. (Engineer) ORG: Hoscow "Order of Lenin" Power Engineering Institute (Moskovskiy ordena Lenina energeticheskiy institut) TITLE: Effect of the departure angle on the characteristics of radially expanding turbine blading SOURCE: IVUZ. Energetika, no. 10, 1965, 56-62 TOPIC TAGS: turbine blade, flow angle, turbine design ABSTRACT: The characteristics of radial turbine blading with a d/l ratio of 2.5 were studied in subsonic air streams at departure angles of 9, 12, 15 and 180. The wheels studied were made up of 30 vanes with a height I=100 mm and identical geometric characteristics in all cases with the exception of the departure angle. The flow parameters were measured in front of and behind the blading. Measurements were made at ten points between blades, in 15-25 sections along the height of the blade and at distances s=5, 15, 25 mm from the outlet edge of the blades which corresponds to Card 1/3 UDC: 621.165

1856U-66	
CO NR: AP6006428	
=z/b=0.125, 0.375, 0.625. The resultant data were used for determining the distri-	
12 12 12 Entrainmentage with respect to the neight of the bidding of the	体を
reaking pressure p_{01} , static pressure p_1 , and flow departure angles a_1 and a_2 in the meridian direction. Angle a_1 is measured between the projection of velocity a_1	15-
and a is measured between the vector or velocity of and its projection on	
lu, angle u2 lo medical surface. It was found that an increment in the effective angle of eparture increases the difference between the static pressures at the periphery and	
ook of the blading due to a reduction in energy losses and a corresponding increase	
- the enteries of the departing air at the root section. Measurements or departure	
ngles al show that they are greater than the effective departure angles. When the ffective departure angle is increased, the difference between the average value of	
La manufaction and the effective angles is reduced. Which may also be explained	
dustion in change losses and less redistribution in the rate or air rick	
ith respect to height. The average values of the meridian angles with respect to lading height (α_2) are a linear function of the effective departure angle:	
α ₂) av =0° for z=0.125 at an effective departure angle of 15°. At smaller effective	
eparture angles; the average values of α_2 are positive, which corresponds to de-	
eparture angles; the average values of u2 are positive; which torresponds to us	15.57 15.57
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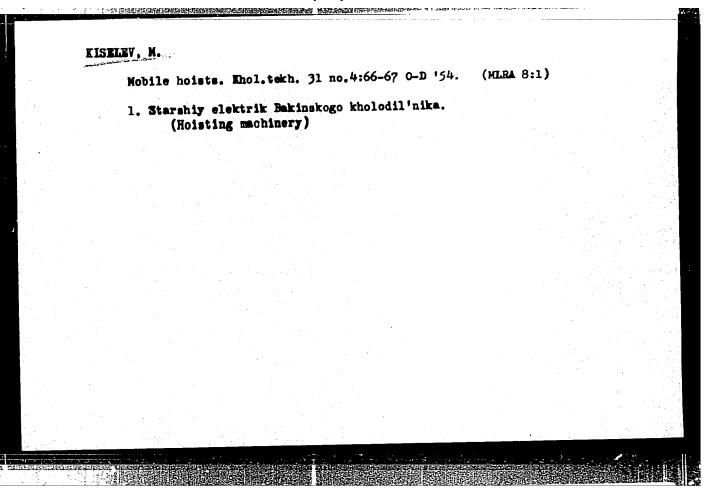
flection of the line of flow from the root to the periphery. This is due both to a high degree of twisting in the stream and to the highly developed region of energy losses in the lower half of the turbine blading. The effective departure angle has a considerable effect on the distribution of energy losses, particularly in the effective angle of departure, especially at great distances from the vanes. Experimental data show a predetached flow and extremely high energy losses in radial-than 15°, Orig. art. has: 7 figures						
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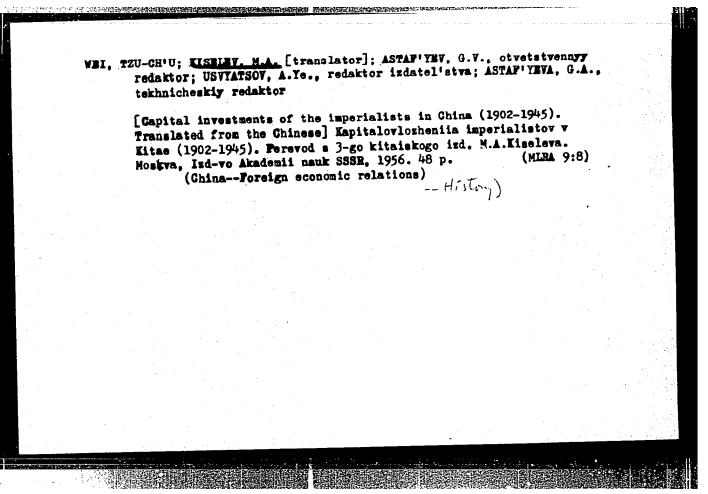
KISELEV, L.Ye.; SHESTOVA, L.M., red. [The North opens up its wealth; from the history of the industrial development of the Soviet Far North] Sever raskryvaet bogatstva; iz istorii promyshlennogo razvitiia

Sovetskogo Krainego Severa. Moskva, Izd-vo "Mysl'," 1964. (MIRA 17:6)

108 p.



APPROVED FOR RELEASE: 09/17/2001 CIA-RDP86-00513R000722810005-6"



و المنظمة المن	A book on the foreign trade of the Chinese People's Republic ("Basic data on the foreign trade of the Chinese People's Republic" [in Chinese] by Ch'i Hsiao-ssu. Reviewed by M. Kiselev). Vnesh. torg. 29 no.4:30-31 159. (MIRA 12:6)		
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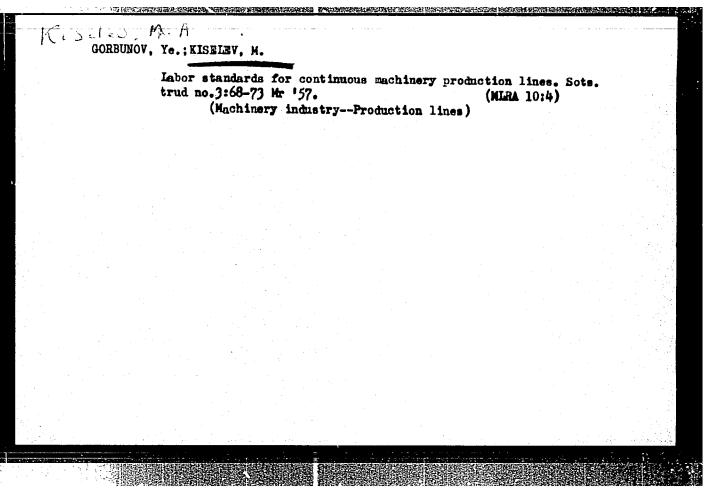
KINELY, M. A. (Engr) -- "Utilization of Experience in the Acclimatization of Varieties of Trees and Bushes of Orlovskoya Chlast in Shelter Belt Planting." Sub 5 Jan 53, Roscow Forestry Engineering Inst. (Dissertation for the Degree of Candidate in Agricultural Sciences).

So: Vechernaya Hoskva January-December 1952

DERYUGIN, Sergey Matveyevich; KISELEV, M.A., retsensent; ZAYTSEVA, T.M., red.;
KOGAN, V.V., tekhn.red.

[Operation and maintenance of spinning machinery; comb spinning of thin wool] Ustroistvo i obslushivante priadil'nykh mashin;
grebennoe priedenie tonkoi shertsi. Moskva, Gos.nauchno-tekhn.
izd-vo lit-ry po legkoi promyshl., 1957. 150 p. (MIRA 11:1)

(Spinning machinery)



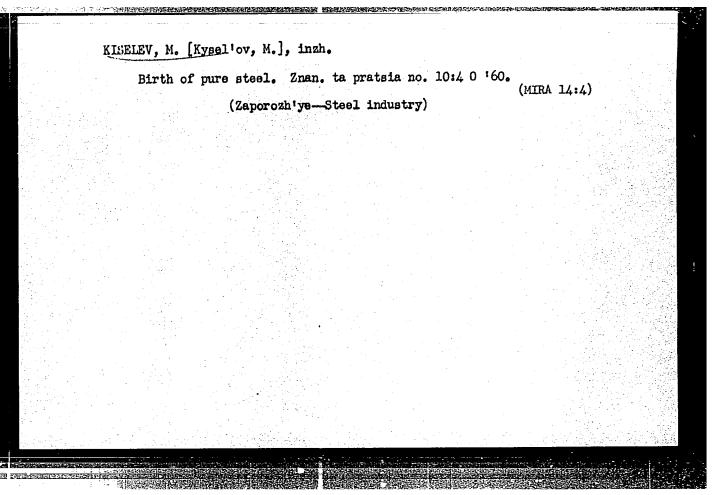
KOVALEV, F.L., kend.tekhn.nauk, red.; GAMBURG, Ya.Yu., retsenzent; FORMAL'SKIY, M.I., retsenzent; KISELEY, M.A., retsenzent; PLEMYANNIKOV, M.N., red.; SUKULOVA, V.Ye., red. EIOZNOV, A.G., red.; KNAKNIN, M.T., tekhn.red.

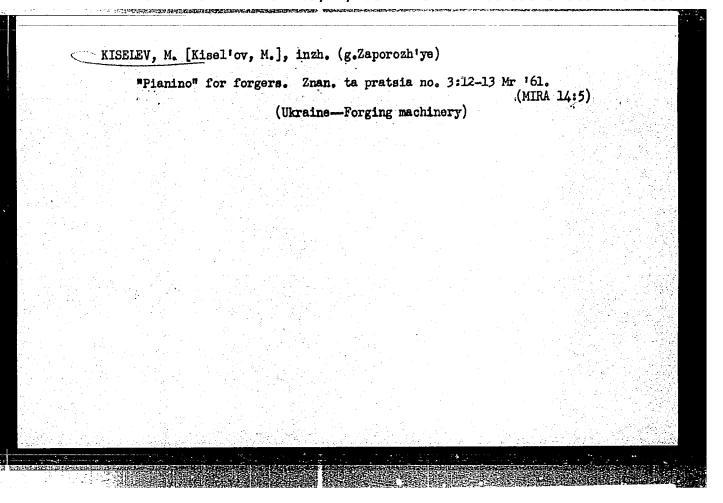
[Manual on wool spinning] Spravochnik po sherstoprisdeniiu. Pod red. F.L.Kovaleva. Izd.2., perer. i dop. Moskva, Izd-vo nauchno-tekhn.lit-ry RSFSR, 1960. 785 p.

(MIRA 13:12)

1. Moscow. TSentral'nyy nauchno-issledovatel'skiy institut sherstyanoy promyshlennosti.

(Woolen and worsted spinning)





KESELEV, M. F.

Cand Geolog-Minerog Sci

Dissertation: "Swelling of Grounds Due to Freezing and Its Effect on the Stability of Foundations."

26 May 49

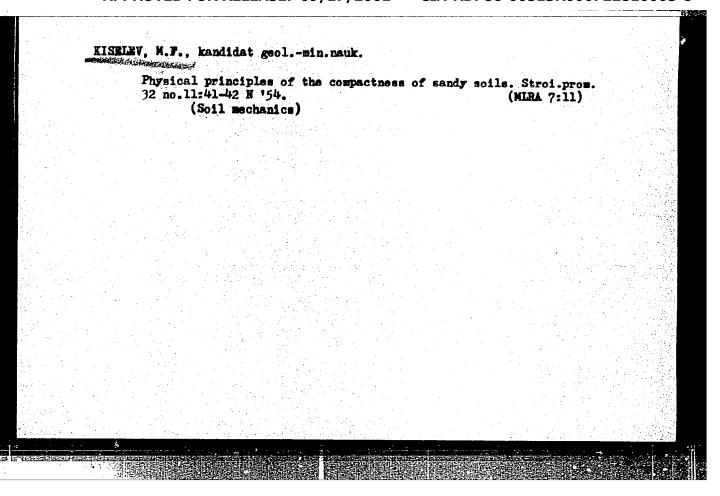
Moscow Order of Lenin State V imeni M. V. Lomonosov

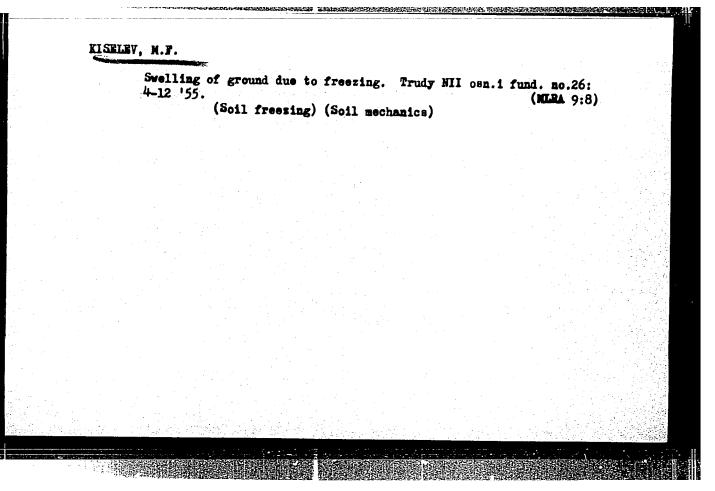
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KISELEV, N. F.

"Swelling of Counds Due to Freezing and Its Effect on the Stability of Foundations."
Thesis for degree of Cand. Ceological - Mineralogical Sci. Sub 26 May 49, Moscow Order of Lenin State U ineni M. V. Lomonosov

Summary 82, 18 Dec 52, Dissertations Presented for Degrees in Science and Engineering in Moscow in 1949. From Vechernyaya Moskva. Jan-Dec 1949





KISELEV, M.F.; GOLUBENKOVA, L.A., red.izd-vs; ELIKINA, E.M., tekhn.red.

[Calculating the setting of foundations on thawing soil beds]

K raschetu osadok fundamentov na ottaivaiushchikh gruntakh-osnovaniiakh. Moskva, Gos. izd-vo lit-ry po atroit. i arkhit., 1957.

39 p.

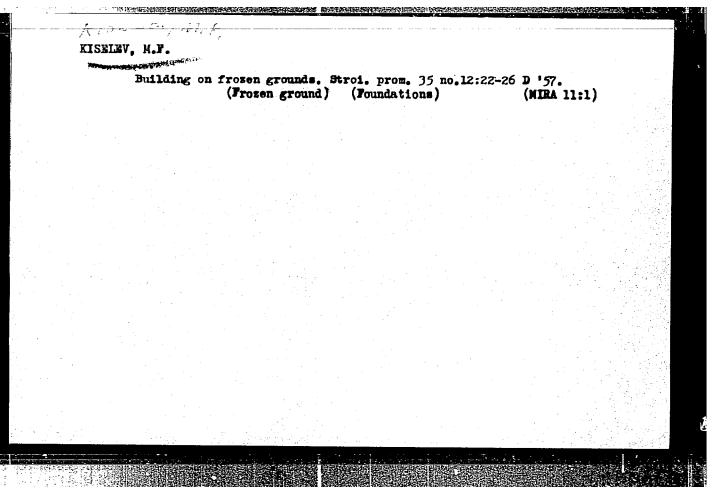
(Foundations) (Soil mechanics)

KISELEY, M.F., kandidat geologo-mineralogicheskikh nauk.

Evaluating suitableness of frosen grounds for laying foundations.

Stroi.pros.35 no.2:37-39 F '57. (MEA 10:3)

(Foundations) (Frosen ground)



KISELEV, M.F.

14(10)

PHASE I BOOK EXPLOITATION

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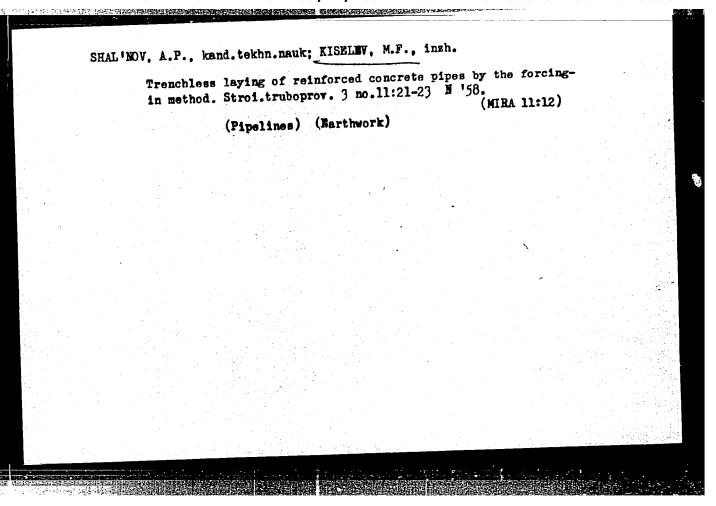
Akademiya stroitel'stva 1 arkhitektury SSER. Institut osnovaniy 1 podsemnykh soorusheniy

Instruktsiya po opredeleniyu raschetnoy glubiny ottaivaniya merslykh gruntov v osnovanii sooruzheniyi po opredeleniyu raschetnykh teplofisicheskikh koeffitsiyentov gruntov (Instructions for Estimating the Depth of Fermane frost Thaving When Laying Building Foundations and for Estimating the Thermophysical. Coefficients of Soils) Moscow, Gosstroyizdat, 1958. 16 p. 5,000 copies printed.

Ed.; M.F. Kiselev., Candidate of Geological and Mineralogical Sciences; Ed. of Publishing House: A.P. Mumits.; Tech. Ed.: L.M. Solntseva.

PURPOSE: This booklet is intended for civil engineers and builders, particularly those encountering permafrost conditions.

COVERAGE: These instructions include methods for estimating the depth to which permafrost thaws when the foundations for buildings and other structures are laid. A method of computing the thermophysical characteristics of soils without conducting field or laboratory tests requiring special equipment is also included. Test situations and examples are given to illustrate the Card 1/2



APPROVED FOR RELEASE: 09/17/2001 CIA-RDP86-00513R000722810005-6"

KISELEV. M.F., kend.geologo-mineral.nauk; KHAVIN, B.K., red.izd-ve; SOLETSEVA, L.N., tekhn.red.

[Instructions for organizing and conducting observations of changes in the water and temperature in permanently frozen ground for the purpose of building foundations]
Ukasaniia po organizatsii i vedeniiu nabliudenii sa ismeneniem vodno-temperaturnogo rezhima vechnomerslykh gruntov dlia tselei fundamentostroeniia. Moskva, Gos.izd-vo lit-ry po stroit., arkhit, i stroit.materialam, 1959. 26 p.

1. Akademiya stroitel'stva i arkhitektury SSSR. Institut osnovaniy i podsemykh soorusheniy.

(Prosen ground)

KISELEV, M. F.: Doc Tech Sci (diss) -- "The theory of compressibility of melting soil". Moscow, 1959. 33 pp (Acad Construction and Architecture USSR, Sci Res Inst of Foundations and Underground Structures), 150 copies (KL, No 13, 1959, 103)

K15ELEV, M.F. 14(10); 3(5) P.Z.

PHASE I BOOK EXPLOITATION

sov/2843

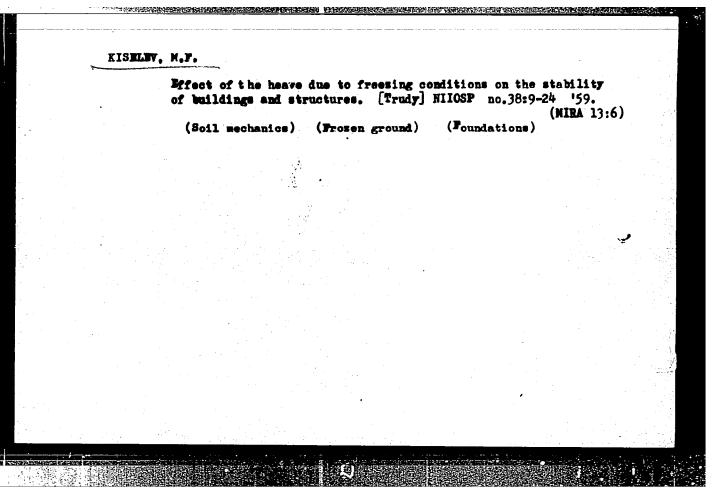
- Soveshchaniye po ratsional'nym sposobam fundamentostroyeniya na vechnomerzlykh gruntakh
- Trudy... (Transactions of the Conference on Efficient Methods of Building Foundations on Permafrost Soils) Moscow, Gosstroyizdat, 1959. 131 p. Errata slip inserted. 1,200 copies printed.
- Ed. of Publishing House: N. M. Borshchevskaya; Tech. Ed.: Ye. L. Temkina.
- PURPOSE: This book is intended for construction engineers, industrial planners and builders.
- COVERAGE: This book contains reports originally read in Vorkuta in 1958 on experience gained in planning and building foundations in permafrost regions of the USSR. The reports were prepared for publication in the NIIOSP (Scientific Research Institute for the Study of Foundations and Underground Structures). The Untroduction was written by Professor V. G. Bulychev. No references are given.

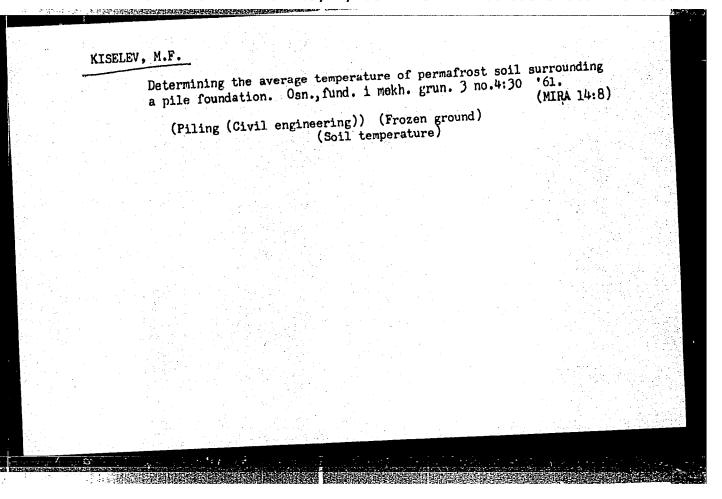
card 1/4

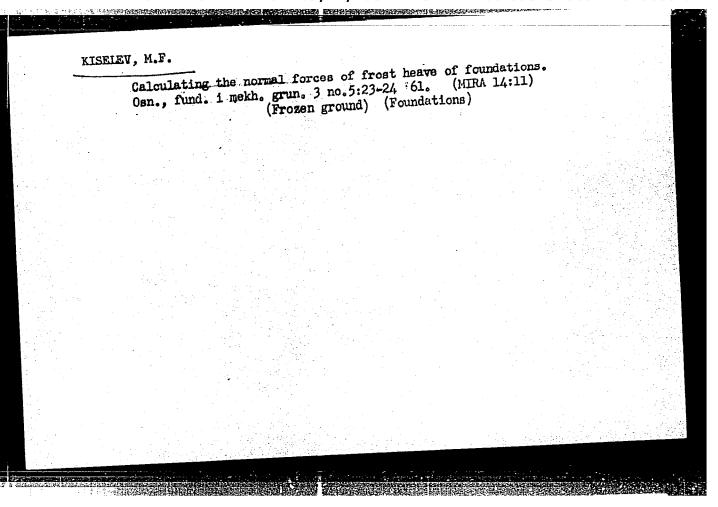
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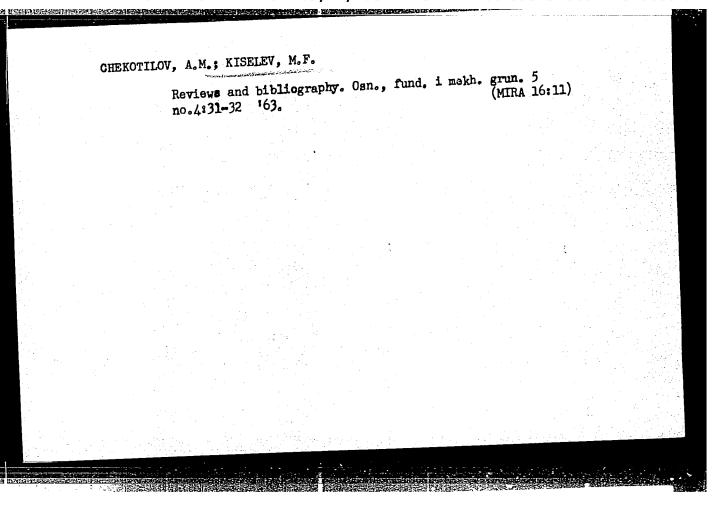
(Foundations) (Frozen ground)	Plan for new norms and technical specifications for designing natural foundations for buildings and structures to be built on permafrost. Osn., fund. i mekh.grun. no.3:16-18 '59. (HIRA 12:8)						
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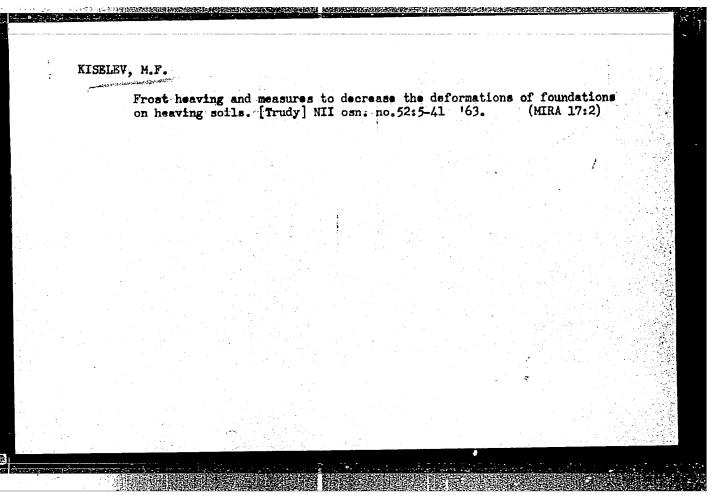


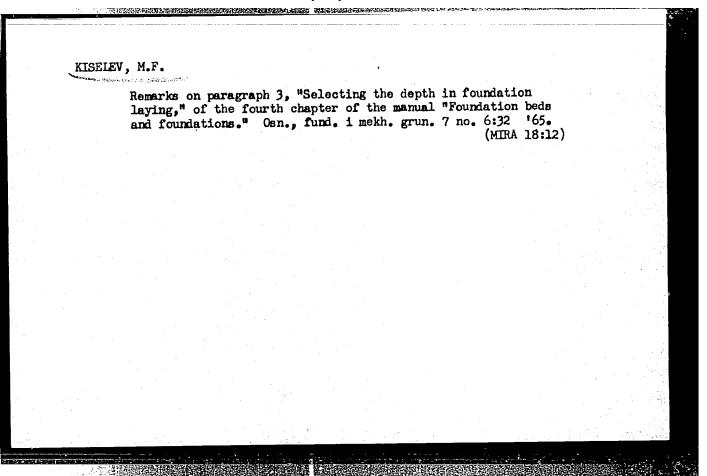
KISELEV, M.F., elektromekhanik; @IMBURG, D.M.

Discussion of the article "Instructions should be reviewed."

Avtom., telem. i sviaz' 7 no.ll;37-39 N '63. (MIRA 16:12)

1. Liskinakaya distantsiya signalizatsii i svyazi Yugo-Yostochnoy dorogi (for Kiselev). 2. Nachal'nik otdela signalizatsii, tsentralizatsii i blokirovki sluzhby signalizatsii i svyazi Severnoy dorogi (for Gumburg).





DZHAVAKHYAN, Tigran Vaganovich, inshener; KISELEV, Mikhail Grigor yevich, inshener; GALAHOVA, M.S., inshener, redaktor; YUDZOH, D.M., tekhnicheskiy redaktor.

[Work practice of departments handling automatic train stops in locomotive repair shops] Opyt raboty sakhov avtostopov lokomotivnykh depo. Moskva, Gos.transp.shel-dor. isd-vo, 1955. 86 p. (Locomotives--Repairs) (MLRA 8:11)

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I.L., red.; KANEVSKATA, M.D., red.; FAINSHMIDT, F.Ya., tekhn.red.

[What everyone should know about antiaircraft defense in the cities] Chto nado snat' naseleniu o protivovozdushnoi oborome gorodov. Moskva, Izd-vo DOSAAF, 1959. 39 p. (MIRA 13:2)

(Air defenses)

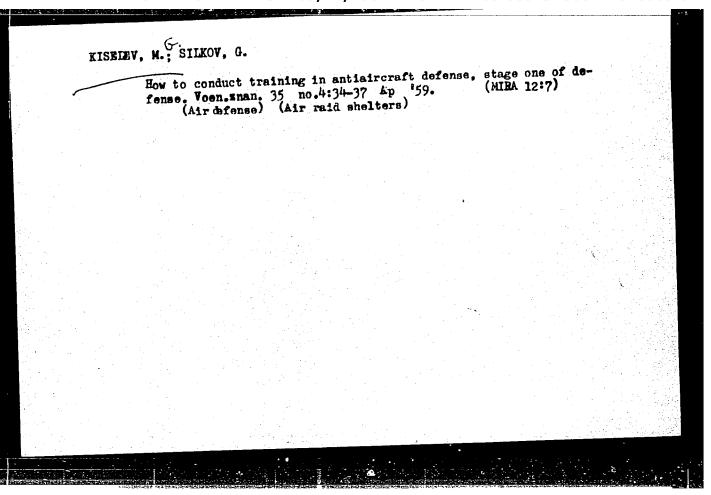
BABKIN, I.A.; VELYUGO, V.M.; DIVAKOV, P.D.; ZAPOL'SKIY, G.N.; KIPRIYAN,
K.M.; KISHAY, M.C.; KORABLEV, M.D.; SILKOV, G.A.; SMORODIN, I.Ya.;
KAREYSKAYA, M.D., red.; GERASIMOVA, V.N., teknn.red.

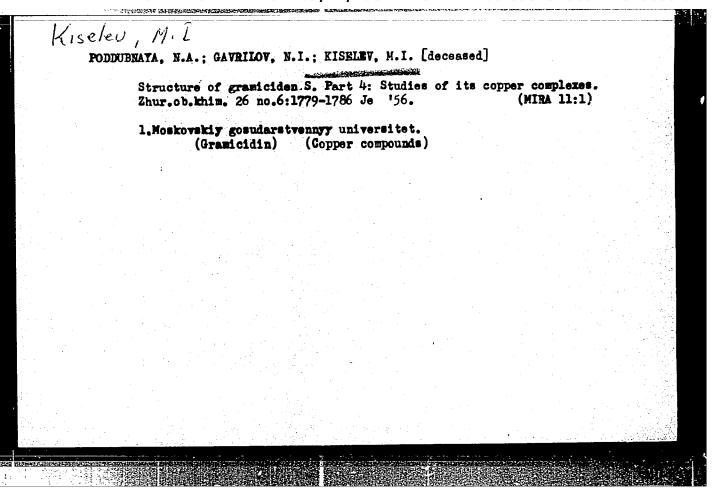
[Manual for training and testing for a first-class rating in the organization "Ready for Antiaircraft Defense."] Uchebno-meto-dicheskoe posobie po provedenilu trenirovok i priemu norm "Gotov k PVO" 1-i stupeni. Moskva, Izd-vo DOSAAF, 1959. 110 pe.

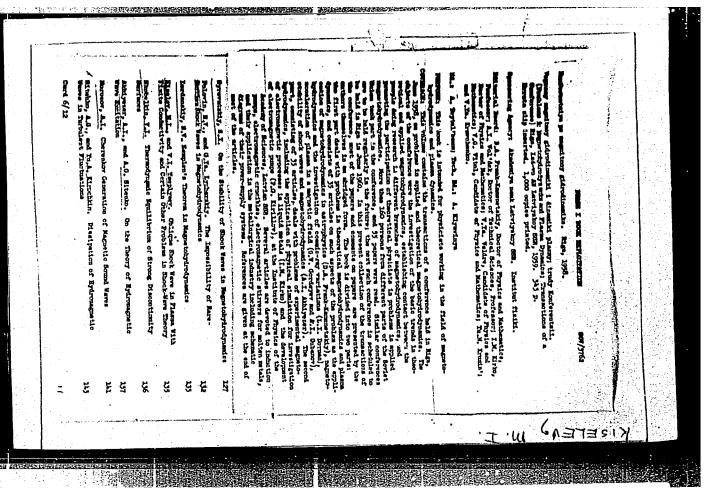
(MIREA-12:5)

1. Vsesoyusnoye dobrovol'noye obshchestvo sodeystviya armii, aviatsii i flotu.

(Civil defense)







SOV/56-34-6-29/51 Tseplyayev, V. I. Kiselev, M. I., AUTHORS:

Inclined Shock Waves in a Plasma With Finite Conductivity TITLE:

(Naklonnyye udarnyye volny v plazme s konechnoy provodi-

most'yu)

Zhurnal eksperimental'noy i teoreticheskoy fiziki, 1958, PERIODICAL:

Vol. 34, Nr 6, pp. 1605-1609 (USSR)

This paper investigates the structure of the front of an inclined shock wave for arbitrary orientations of the field ABSTRACT:

before the front in a plasma with finite conductivity. This plasma is assumed to have a constant and isotropic conductivity o which is high enough for the displacement current to be neglected. The authors obtain the conditions for the possibility of neglecting the kinematic viscosity V and the thermal conductivity due to the electrons \varkappa . (ϑ and \varkappa are neglected with respect to the magnetic viscosity ν in the system of the equations of magnetic hydrodynamics). This condition

 $v_m\gg v$ is specialized also for a special case. Then the

authors give the particular integrals of the equations of

magnetic hydrodynamics. The second part of this paper deals Card 1/2

sov/56-34-6-29/51

Inclined Shock Waves in a Plasma With Finite Conductivity

with the structure of an inclined shock wave in a plasma with finite conductivity. The above mentioned particular integrals of motion are specialized to this case. One integral is computed numerically and an expression is obtained for the breadth of the front. The last part of this paper calculates the limit angle of the propagation of the inclined shock wave in a plasma with infinite conductivity. The boundary conditions are given also for this case. In the presence of a magnetic field, the above mentioned limit angle is larger than in the case where there is no magnetic field. The author thanks K. P. Stanyukovich who proposed this problem and was constantly interested in this paper. There are 2 figures and 4 references, 2 of which are Soviet.

ASSOCIATION: Moskovskiy gosudarstvennyy universitet (Moscow State University)

SUBMITTED: January 20, 1958

Card 2/2

10 (4) AUTHOR:

Kiselev, M. I.

SOV/20-126-3-18/69

TITLE:

On the Calculation of Shock Waves in Magnetic Hydrodynamics (K raschetu udarnykh voln v magnitnoy gidrodinamike)

PERIODICAL:

Doklady Akademii nauk SSSR, 1959, Vol 126, Nr 3,

pp 524-527 (USSR)

ABSTRACT:

The calculation of the parameters of a magneto-hydrodynamic shock wave in perfect gases meets with considerable difficulties, because the algebraic equations are of the third and higher degrees. Thus, the boundary condition in the front of a perpendicular magneto-hydrodynamic shock wave is a cubic equation. This equation may, however, be reduced to a quadratic equation providing that it contains all quantities with the index 1 (characterizing the parameters of the medium before the front) as trivial solutions. The system of equations (1) shows the already reduced equations for the velocity of the medium, for its density, and for the magnetic field. The Prandtl-relation and the Mach number are then briefly dealt with. The same is then done with the adiabatic curves and the propagation of the wave front. Finally, inclined magneto-hydrodynamic shock waves are investigated. Figure 1 shows the

Card 1/2

On the Calculation of Shock Waves in Magnetic Hydrodynamics

SOV/20-126-3-18/69

family of impact polars, from which it follows that the angle of inclined fronts of the shock waves increases, and that the angle of rotation of the flow is reduced. Finally, the geometric interpretation of the onedimensional hydrodynamic impact-transition is dealt with and a diagram is given. The author thanks Professor K. P. Stanyukovich for discussions and advice, and G. S. Golitsyn for looking over the manuscript. There are 2 figures and 6 references, 4 of which

are Soviet.

ASSOCIATION:

Moskovskiy gosudarstvennyy universitet im. M. V. Lomonosova

(Moscow State University imeni M. V. Lomonosov)

PRESENTED:

February 13, 1959, by N. N. Bogolyubov, Academician

SUBMITTED:

February 10, 1959

Card 2/2

10.2000 A AUTHORS:

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Kiselev, M. I., Kolosnitsyn, N. I.

S/020/60/131/04/016/073 B013/B007

TITLE:

Calculation of Inclined Shock Waves in Magnetic Gas Dynamics

Doklady Akademii nauk SSSR, 1960, Vol 131, Nr 4, pp 773-775 (USSR)

PERIODICAL: TEXT: The tangential component of the magnetic field behind the front

of an inclined shock wave can be calculated from the cubic equation (1):

of an inclined shock wave can be calculated from
$$h_{2y}^3 + h_{1y} \left[1 - (2-k) \left(1 - \frac{u_{1x}^2}{v_{1x}^2} \right) \right] h_{2y}^2 + \varrho_1 \left(1 - \frac{u_{1x}^2}{v_{1x}^2} \right) \left(v_{1x}^2 - a_{1Mx}^2 \right) (k+1) h_{2y}^2 - \frac{u_{1x}^2}{v_{1x}^2} \right)$$

. The intensity of the compression shock is $- (k+1) \nabla_{1x}^{2} h_{1y} Q_{1} \left(1 - \frac{u_{1x}^{2}}{V_{1x}^{2}} \right) = 0$

determined by the Mach number $M_{\text{Mach}} = u_{\text{X}}/v_{\text{X}}$ and by the parameter $v_{\text{X}}^2 - a_{\text{Mach}}^2$ The amount of the velocity component uny which is parallel to the front has no influence on the compression shock of the field. By means of linear and broken, linear substitutions it is possible to obtain cubic equations from equation (1),

Card 1/3

Calculation of Inclined Shock Waves in Magnetic Gas S/020/60/131/04/016/073

Bynamics S/020/60/131/04/016/073

which are used to determine the velocities u_{2x} , u_{2y} and the density ϱ_2 . The accelerated shock waves are described by that branch of the roots of these cubic equations which has a positive real part. These shock waves pass over into the inclined shock waves of non-magnetic gas dynamics if the field strength tends to zero. The delayed shock waves (which vanish in perpendicular fields) are described by the branch of roots having a negative real part. With Hach = 1 the afore-mentioned equation (1) is solved by the rotational discontinuity $h_{2y} = -h_{1y}$. Figure 1 shows u_{2y} as a function of u_{2x} . The energy of the magnetic field is higher than or equal to the internal and kinetic energy of the gas. Whereas the field strength behind the front varies in a monotone manner, the dependence of u on u has the character of a hump the peak of which corresponds to a certain "resonant slope" of the magnetic field. In solving the problem of a piston, the collisions between the inclined shock waves and their reflections from the wall are the most interesting kinds of behavior in the neighborhood of the "resonant slope" of field strength. These considerations also permit an explanation of the prominences in the neighborhood of sunspots. By means of the results obtained here it is also possible to

Card 2/3

Calculation of Inclined Shock Waves in Magnetic Gas Dynamics

13/020/60/131/04/016/073 18013/8007

set up a nomogram for calculating the flow of a conducting supersonic flow around a wedge, which is explained here in detail. In the case of a perpendicular wave it is easily possible to determine the discontinuous change in temperature and entropy in an explicit manner. This discontinuous change decreases with increasing field strength with otherwise equal parameters. The

unsteadiness becomes weak with the field strength $h = \sqrt{(u^2 - c^2)}$. The authors thank Professor K. P. Stanyukovich for a discussion of the results obtained in this paper. There are 4 figures and 4 references, 3 of which are Soviet.

ASSOCIATION: Moskovskiy gosudarstvennyy universitet im. M. V. Lomonosova (Moscow State University imeni M. V. Lomonosov)

PRESENTED: December 11, 1959, by N. N. Bogolyubov, Academician

SUBMITTED: December 10, 1959

Card 3/3

S/188/62/000/001/005/008 B125/B138

AUTHOR:

Kiselev. M. I.

TITLE:

Theory of the magnetogasdynamic generator

PERIODICAL: Moscow. Universitet. Vestnik. Seriya III.

astronomiya, no. 1, 1962, 63-68

TEXT: The author studies the influence of the channel profile on the efficiency of the thermal to electric energy conversion and on the dimensions of a magnetogasdynamic generator for the two plane flows in cylindrical coordinates I) $\{v_r, H_z, j_r, II\}$ $\{v_r, H_z, j_z\}$ and the spatial in spherical coordinates. The set of equations

3h273 S/188/62/000/001/005/008 B125/B138

Theory of the magnetogasdynamic ...

$$\frac{dx}{dt} = A\psi^{3} \frac{(\gamma - 1)\left(y - \frac{C_{1,2}}{t^{m_{\psi}}}\right)^{3} \frac{1}{t^{N}} - \frac{N\gamma}{A} \frac{xy}{t^{N+1}} \frac{1}{\psi^{3}} + \left(y - \frac{C_{1,2}}{t^{m_{\psi}}}\right)x}{y|t^{N} - x_{1}B|A}$$

$$\frac{dy}{dt} = -B\psi^{3} \frac{(\gamma - 1)\left(y - \frac{C_{1,2}}{t^{m_{\psi}}}\right)^{3} - \frac{N\gamma}{A} \frac{xy}{t} \frac{1}{\psi^{3}} + \left(y - \frac{C_{1,2}}{t^{m_{\psi}}}\right)^{y}}{y|t^{N} - x_{1}B|A}$$

$$\frac{dz}{dt} = B\psi^{3} \frac{(\gamma - 1)\left(y - \frac{C_{1,2}}{t^{m_{\psi}}}\right)^{3} \frac{1}{y^{3}t^{N}} - \frac{N}{B} \frac{1}{t^{2N+1}} \frac{1}{\psi^{3}} + \left(y - \frac{C_{1,2}}{t^{m_{\psi}}}\right)^{x}}{y|t^{N} - x_{1}B|A}$$

$$\frac{\partial}{\partial t} \frac{1}{t^{n}} \frac{\partial}{\partial t} (t^{n}\zeta) = \operatorname{Re}_{M}\psi\left(y - \frac{C_{1,2}}{t^{m_{\psi}}}\right),$$

with $\psi = +$; n = 0, $\beta = -1$ for I), n = 1, $\beta = -1$ for II) and III) which holds if the equipotential surfaces of the electric field induced in the gas flow lie on the conducting wall of the channel, has the first integral yzt N = +1. The minus sign is for the convergent and the plus sign for the divergent duct flows. The variables $x = p/p_0$, $y = u/u_0$, Card 2/8/5

e34273 S/188/62/000/001/005/008 B125/B138

Theory of the magnetogandynamic ...

t = f/f_0 , t = r/r_0 , ψ = (H+h)/H₀ denote the dimensionless quantities pressure, velocity, density, radius, and magnetic field strength, f is the dimensionless inducing field, (is the induced field and ψ is the sum of these fields. In the current density f = f

$$(\gamma - 1) \left(y - \frac{c_{1,2}}{t^m \psi} \right)^2 - \frac{N y^2}{B t^{N+1}} + \left(y - \frac{c_{1,2}}{t^m \psi} \right) \frac{y}{t^N} = 0.$$
 (2)

holds for I, II, and III. In convergent ducts, and divergent ducts with

3\1273 S/188/62/000/001/005/008 B125/B138

Theory of the magnetogasdynamic ...

quite small divergence angles the subsonic flow releases energy on cooling. In wider-angle diffusors: the gas is decelerated on flowing out and the energy conversion is "suppressed" by the geometrical action. In and the energy conversion is "suppressed" by the geometrical action. In converging ducts and small-angle diverging ducts supersonic flows strike converging ducts and small-angle diverging ducts supersonic flows strike converging ducts and small-angle diverging ducts supersonic flows strike converging to the energy transformation in the subsonic region raising the efficiency of the energy transformation in the subsonic region also increase the kinetic energy of the flow and reduce its internal also increase the kinetic energy of the flow and reduce its internal energy. The current passing through the load is equal to the integral of the current density at the conducting wall with the area and the conducting wall with the area.

efficiency of the generator is

$$\eta = \frac{IV}{\sum_{\mathbf{b}_{\mathbf{x}}\rho_{0}} u_{\mathbf{0}} \left(\frac{\gamma \rho_{0}}{(\gamma - 1)\rho_{0}} + \frac{u_{0}^{2}}{2}\right)} = \operatorname{Re}_{\mathbf{M}} \frac{S_{\mathbf{b}_{\mathbf{M}}}}{S_{\mathbf{r}}} I_{1,2,3} = \frac{2C_{1,2}B}{2 - 1_{1,2,3}}$$

$$= \frac{2}{\gamma - 1} \frac{1}{M_{0}^{2}} + 1$$
(4)

card 4/85

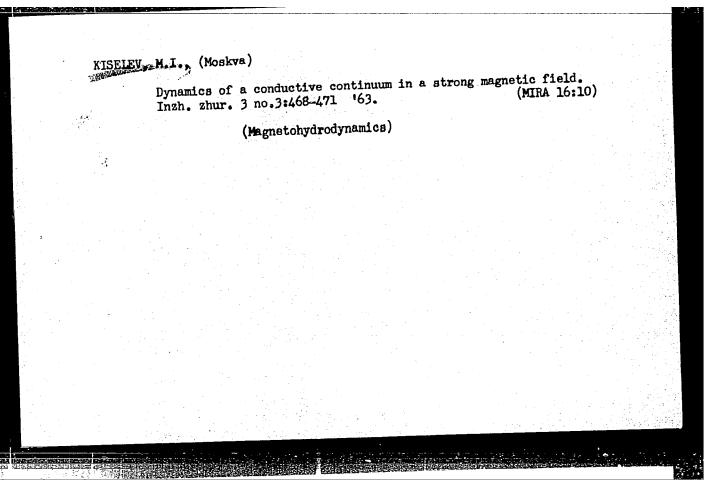
34273 s/188/62/000/001/005/008 B125/B138

Theory of the magnetogasdynamic ...

where S_r and S_{em} are the Poynting vectors of the gas and the induced field at the channel opening. With $Re_M \ll 1$, H = const, V = 5/3 the efficiency is about two thirds of the initial energy flux in the range $2 \leq M \leq 6$. In the range $Re_M > 1$ the effect of the profile shape increases and the effect of the Joulean heat can be neglected in first approximation. Optimum efficiency can only be calculated by means of more realistic models (dissipation in boundary layers, effect of space charge) etc. Professor (dissipation in boundary layers, effect of space charge) etc. Professor (R. P. Stanyukovich, Professor V. L. Granovskiy are thanked for discussing the results; G. S. Aravin, N. I. Kolosnitsyn, V. P. Shevelev, and I. I. Nochevkina for numerical computations. The Monograph of Bulis L. A. Nochevkina for numerical computations. The menograph of Bulis L. A. Gazovaya dinamika. M., 1949 is quoted. There are 3 figures and 5 cazovaya dinamika. M., 1949 is quoted. There are 3 figures and 5 cazovaya publication reads as follows: Neuringevr L. J. Fluid Mech., 7. No. 2, 287, 1960.

ASSOCIATION: Kafedra statisticheskoy fiziki i mekhaniki (Department of Statistical Physics and Mechanics)

card 5/85



MT(m)/MA(h)/MP(b)/MP(t) UR/0286/65/000/014/0059/0059 ACCESSION NR: AP5021985 621.365.69.013 : 534.8.004 AUTHOR: Kiselev, M. I.; Makarov, A. M.; Feoktistov, V. A. TITIE: A method for heating electrically conductive materials. Class 21, SOURCE: Byulleten' izobreteniy i tovarnykh znakov, no. 14, 1965, 59 TOPIC TACS: magnetic field, electric conductivity, heating, magnetic effect, ultrasonic effect, ul/rasonic vibration ABSTRACT: This Author's Certificate introduces: 1. A method for using a magnetic field to reat electrically conductive materials. The heating is intensified by excitation of ultrasonic oscillations in the material peing heated in a direction per pendicular to the lines of force of the magnetic field. 2. A modification of this method in which residual stresses are eliminated in the material being heated by removing the magnetic field before killing the ultrasonic oscillations. ASSOCIATION: none SUB CODE: ENCE: 00 SUBMITTED: 01Nov63 OTHER: 000 NO REF SOV: 000 Card 1/140

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CIA-RDP86-00513R000722810005-6

L 56552-65 EWP(m)/EWG(τ)/EWT(1)/FS(τ)-3/EEC(a)/EEC(τ)/EEC(τ)/EWA(d) Pe-5/

Pg-4/Po-4/Pq-4 GW

ACCESSION NR: AP5015667

UR/0293/65/003/003/0391/0394 629.19:531.38

AUTHORS: Galitskaya, E. B.; Kiselev, M. I.

TITLE: Radiation orientation of cosmic devices

SOURCE: Kosmicheskiye issledovaniya, v. 3, no. 3, 1965, 391-394

TOPIC TAGS: radiation balance, satellite orientation, satellite orientation stability, solar radiation absorption, solar radiation effect, space vehicle design

ABSTRACT: An analysis was made of the basic principles involved in using solar radiation pressure to control the <u>orientation</u> and stability of space craft. This idea was first proposed in 1961; since it is now feasible to build space craft with large enough dimensions to provide the necessary orientating rotational with large enough dimensions to provide the necessary orientating rotational moments, the idea is explored further. The mathematical analysis is on a model consisting of three pairs of flat blades, one lying in each plane of the coordinate consisting of three pairs of flat blades, one lying in each plane of the coordinate consisting of three pairs of flat blades, one lying in each plane of the coordinate system. The orientation between the moving reference system attached to the observation ject and the earth's first coordinate system is established on the basis of Euler ject and the earth's first coordinate system is established on the basis of Euler angles. By using matrix transformations, the expressions for the rotational moments are developed. These expressions indicate that light striking a blade

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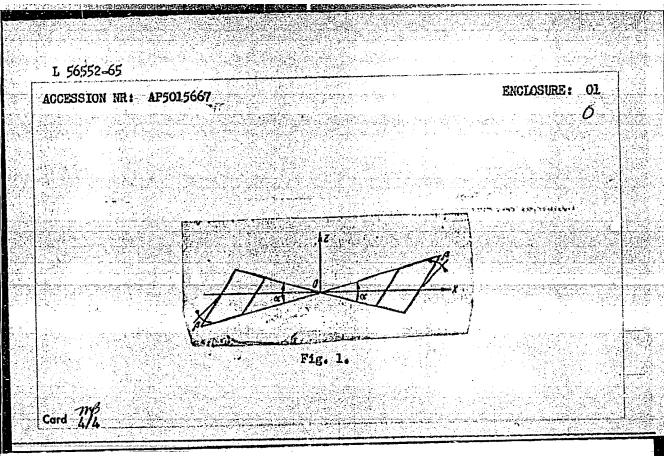
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ACCESSION NR: AF5015667

at an angle provides the necessary controlling turn to orientate the apace craft as desired. It is noted that a "radiation precession" also arises and must be considered in the controlling problem. An investigation of small perturbations of a spherically symmetrical object about the ideal orientation reveals that the ratio of the maximum auxiliary rotational moments to the controlling moments is equal to 3.5% of the basic controlling moments for each degree of deflection from equilibrium. A "radiation propeller" (see Fig. 1 on the Enclosure) was studied to evaluate the magnitude of the rotational moment and the coefficient of utilization of the solar pressure. Two blades with an angular span of lie in the XOY plane. The blades are free to turn to an angle & about an axis lying in the XOY plane. The light strikes along OZ. The rotational moment results from the forced component perpendicular to OZ. The rotational moment, therefore, is 0 with $\beta=0$ or $\eta/2$ and is shown to be a maximum when $\beta \simeq 35$ degrees. With β at 35 degrees, an ideal perfect reflector would give a maximum coefficient of solar pressure utilization of 38.5%. A half white, half black coating of a space craft would harness this solar pressure to provide an orientation with the white (reflecting) side of the space craft to the sun. This would have the additional advantage of improving the space craft's radiation balance. The expression is presented for the condition of stabilization of a black-white sphere provided with "a radiation propeller" in respect to the "radiation precession." The spectral dependency of Card 2/4

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he absorption also was cons A. Skuridin. Orig. art. has	idered. The authors thank : 3 figures and 20 formula	K. P. Stanyukovich and G.	
ASSOCIATION: none			
Submitted: 20May64	ENCL: 01	SUB CODE: NG, SV	
NO REF SOV: 003	OTHER: 005		
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ard 3/4			



ACC NR: AP6019603

SOURCE CODA: UR/0293/66/004/003/0492/0493

AUTHO APPROVED FOR RELEASE: 09/17/2001 CIA-RDP86-00513R000722810005

ORG: none

TITLE: Effect of optic absorption on radiation orientation of spacecraft

SOURCE: Kosmicheskiye issledovaniya, v. 4, no. 3, 1966, 492-493

TOPIC TAGS: space orientation, satellite orientation

ABSTRACT: An orientation control system for spacecraft based on the utilization of solar radiation pressure on a system of control vanes forming a "solar propeller" is considered. The results of a previous article by E. B. Galitskaya and M. I. Kiselev (Kosmich. issled., 3, No. 3, 391, 1965), in which ideal reflection was considered, is extended to include the effects on the radiation rotational device of absorption by the vane material. Expressions are found for the torques acting on the pair of symmetric vanes, which effect control by rotating the spacecraft about the OZ axis and create a secondary torque about the OX axis with oblique incidence of the light. If the incident light is parallel to the OZ axis, the secondary torque Mx vanishes and the control torque is

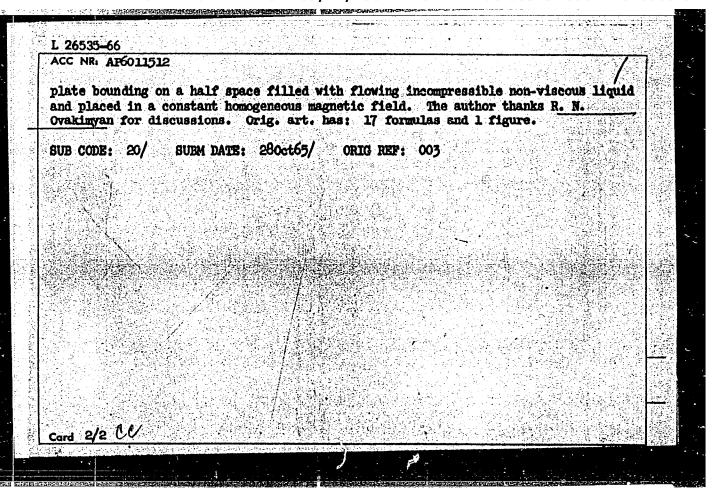
 $M_z = -4B(1-s)\cos^2\mu\sin\mu$,

where the coefficient B is determined by the geometry and dimensions of the vanes and

Card 1/2

VDU: 629.196.3

EWP(m)/EWT(1)/EWT(m)/ETC(m)-6/T/EWA(d)/EWA(1) IJP(c) L 26535-66 ACC NR: AP6011512 SOURCE CODE: UR/0382/66/000/001/0051/0054 AUTHOR: Kiselev, M. I. TITIE: On magneto-elastic flutter SOURCE: Magnitnaya gidrodinamika, no. 1, 1966, 51-54 TOPIC TAGS: magnetohydrodynamics, mid instability, elastic oscillation, or operated from the fill, incomplication flow, including the state of ABSTRACT: The author investigates in the linear approximation the instability of flow of an incompressible non-conducting liquid in a thin-wall elastic conducting tube, placed in a constant homogeneous magnetic field parallel to the tube axis. The induced magnetic field and the viscosity of the liquid are neglected. The purpose of the investigation is to study the spontaneous occurrence of oscillations in current-carrying elements which are cooled by large streams of non-conducting liquid, such as in large-current installations used to produce magnetic fields. The behavior of the cooling tube is described in the approximation of elastic cylindrical shell under the influence of small perturbing radial oscillations. It is shown that there is a limiting flow velocity, beyond which flutter is excited; this limiting velocity is determined by the elastic properties of the shell material, by its radius, by its thickness, and by the density of the liquid. If the radius of the tube is allowed to increase to infinity, the results of the present analysis lead to a solution of the problem of excitation of oscillations in a system made up of an elastic conducting Card 1/2 UDC: 533.95: 538.4



31525-66 ACC NR AP6008832

SOURCE CODE: UR/0294/66/004/001/0087/0091

AUTHOR: Kiseley, M. I. (Moscow); Stanyukovich, K. P. (Moscow)

ORG: None

TITLE: Contribution to the theory of nonlinear skin effect

SOURCE: Teplofizika vysokikh temperatur, v. 4, no. 1, 1966, 87-91

TOPIC TAGS: heat conductivity, skin effect, magnetic field, metal conductivity, metal

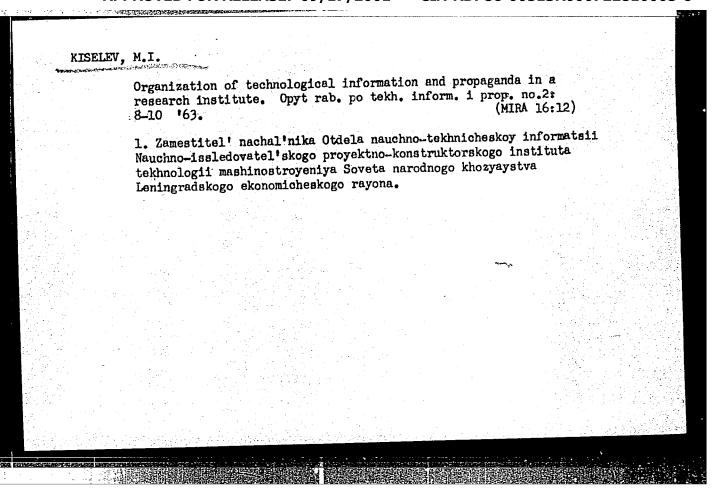
heating

ABSTRACT: The authors investigate some solutions of the nonlinear equation of heat conductivity applicable to the description of skin effect phenomena in highly magnetic fields. Two possible versions of the evaluation of the depth of penetration of an electromagnetic field into a metal are studied for the case when the Joule heat release is so high that it is necessary to take electrical conductivity variations with the temperature into consideration. Such evaluations may prove to be interesting for technical applications in heating metal by means of high frequency currents or high-current pulses, in interpreting data in an electrophysical experiment, and in describing transient processes in the critical point of superconductivity. Orig. art. has: 1 figure and 24 formulas.

SUB CODE: 11,20 / SUBM DATE: 06Jan65 / ORIG REF: 002 / OTH REF: 001

UDC: 538.6:537.29

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ACC NR. AP7001038 SOURCE CODE: UR/0244/66/004/003/0451/0459	- 7
L 07139-67 EWT(1)/EWP(m) IJP(c) ACC NR: AP7001038 SOURCE CODE: UR/0294/66/004/003/0452/0453 KISELEV, M.I. (Mosecu) "Self-Similar Motion in Magnetogasdynamics with Variable Conductivity"	
Moscow, Teplofizika Vysokikh Temperatur (High-Temperature Physics), Vol. 4, No. 3, May-June 1966, pp 452-453	
Abstract: In real processes magnetic conductivity and consequently, "magnetic viscosity" vary together with the thermodynamic parameters of a medium. It is shown that in many cases It is possible to find self-similar motion in a medium that has variable conductivity. The author introduces a system of equations for symmetrical flow and assumes that magnetic viscosity depends on pressure, density, and the magnetic field. Under certain assumptions, the system of equations has self-similar solutions for motion in a conducting medium in a strong field. The limitations of the approach are given, as well as the condition of current distribution in an un-	
perturbed medium for self-similar motion to exist. The author thanks K.P., Stanyukovich for discussing the results, Orig. art. has: 7 formulas. [JPRS: 37,872]	
Stanyukovich for discussing the results, Orig. art. has: 7 formulas. [JPRS: 37,872]	
Stanyukovich for discussing the results, Drig. art. has: 7 formulas. [JPRS: 37,872] ORG: none TOPIC TAGS: magnetogasdynamics, magnetic viscosity SUB CODE: 20 / SURM DATE: 10Apr65 / ORIG REF: 004 /	
Stanyukovich for discussing the results, Orig. art. has: 7 formulas. [JPRS: 37,872] ORG: none TOPIC TAGS: magnetogasdynamics, magnetic viscosity	

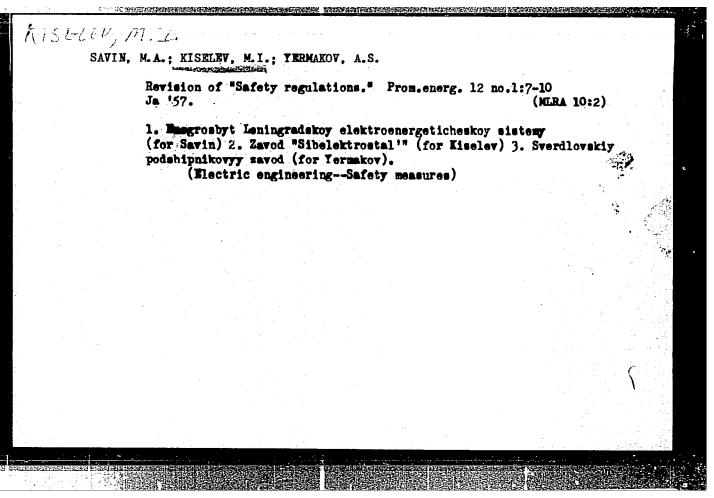


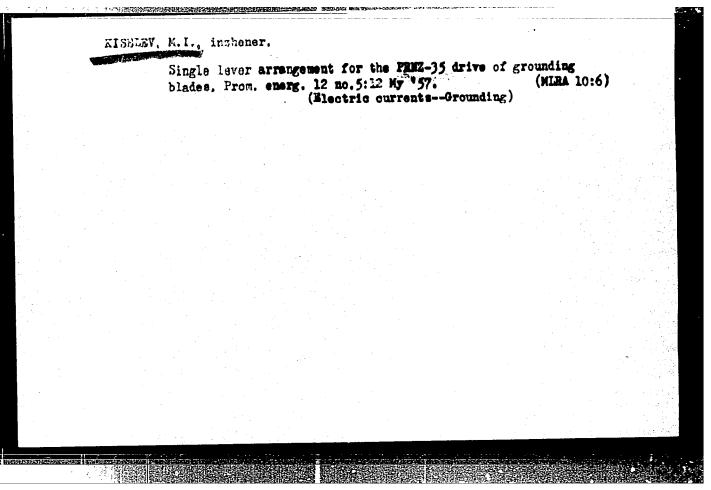
Improving the reliability of PS-10 drives. Prom. energ. 11 no.10:12 (MIRA 9:11) (Electric driving)	KISELEV	M.I., inshener.

Wiseley, M.I., inzhener.

Using VMG-133 circuit breakers in electric arc furnaces.
Prom.energ. 11 no.7:10-11 J1 '56. (MIRA 9:10)

(Electric circuit breakers) (Electric furnaces)





Operation of the VMO-133 cut-out switch in arc furnaces. Pros. energ. 12 no.6:6-8 Je '57. (MIRA 10:7)

1. Krasnoyarskiy zavod (for Kiselev). 2. Gosudarstvennyy politekhnicheskiy institut Tyashproselektroproyekt.

(Blectric cutouts)

KISELEU

AUTHOR: Kiselev, M.I.

94-3-7/26

TITLE:

The Use of Selsyns to Indicate Level (Primeneniye sel'sinov

dlya ukazaniya urovney)

PERIODICAL: Promyshlennaya Energetika, 1958, Vol.13, No.3, pp. 12 - 13 (USSR).

ABSTRACT: Although float-operated limit switches are often used to indicate the level of liquid in tanks, float-operated relays for continuous indication of the liquid level over a wide range are not very satisfactory. However, a pair of selsyns, type CC-404, of 110 V and 0.42 A, were successfully used for this purpose. The rotor of one selsyn was geared to the shaft of the float switch pulley and the rotor of the other to the instrument needle. The only disadvantage of the method is that it requires five wires. The device is reliable and cheap. There is 1 figure.

ASSOCIATION: Sibelektrostal' Works (Zavod "Sibelektrostal'")

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Card 1/1