LYUBIMOV, Georgiy Aleksandrovich; L'UBIMOV, Boris Georgiyevich;
GEYMAN. M.A., nauchn. red.; SHVETSOVA, E.h., ved. red.;
DEL'YANENCO, V.I., tekhn. red.

[Theory and design of axial multistage turbodrill turbines]
Teoriia i raschet osevykh mnogostupenchatykh turbin turboburov. Leningrad, Gostoptekhizdat, 1963. 178 p.

(MI:A 17:2)

6

10(7) AUTHOR:

Lyubimov, G.A.

SOV/55-58-5-6/34

TITLE:

On the Influence of Viscosity and Thermal Conductivity on the Gas Flow Behind a Strongly Curved Shock Wave (O vliyanii vyazkosti i teploprovodnosti na techeniye gaza za sil'no iskriv-

lennoy udarnoy volnoy)

PERIODICAL:

Vestnik Moskovskogo universiteta, Seriya matematiki, mekhaniki, astronomii, fiziki, khimii, 1958 Nr 5, pp 33 - 36 (USSR)

ABSTRACT:

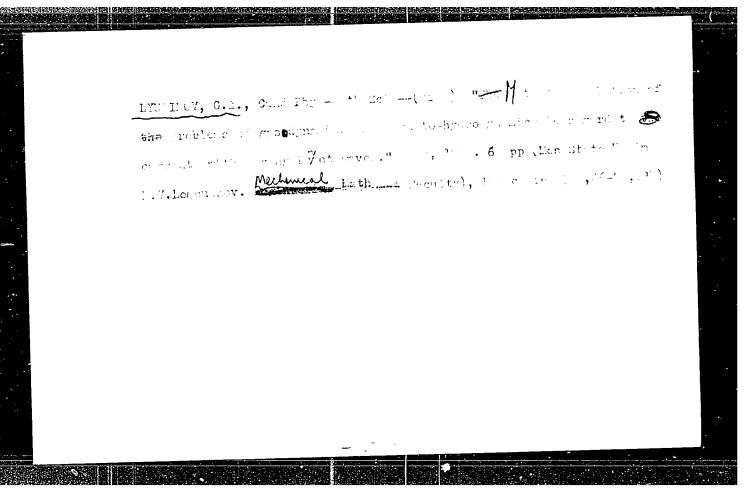
In [Ref 1] L.I. Sedov and others gave formulas for the relations of the braking temperatures and pressures in front of and behind a curved shock wave under consideration of viscosity and heat conduction of the gas, whereby plane and axialsymmetric waves were considered. The author generalizes these

results to shock waves of arbitrary form. There are 2 figures, and 1 Soviet reference.

ASSOCIATION: Kafedra gidromekhaniki (Chair of Hydromechanics)

May 28, 1958 SUBMITTED:

Card 1/1



24 (1)

· AUTHOR:

Lyubimov, G. A.

SOV/ 55-58-6-3/31

TITLE:

On the Compression of a Gas Cylinder by Means of Current (O

szhatii gazovogo tsilindra tokom)

PERIODICAL:

Vestnik Moskovskogo universiteta. Seriya matematiki, mekhaniki, astronomii, fiziki, khimii, 1958, Nr 6, pp 13 - 17 (USSR)

ABSTRACT:

In the present paper the solution of the problem mentioned in the title is carried out. In this investigation - in contrast to other papers (Refs 1,2) - the shock wave which forms due to a magnetic field is taken into account. This magnetic field forms due to the feeding of the gas cylinder with electric corrent. The solution is obtained in the form of a series, developed according to low powers of ξ . In this connection ξ denotes the ratio of the gas density before and behind the shock wave. For the solution first the system of equations (1)

 $\frac{\partial r^2}{\partial t^2} = -r \frac{\partial p}{\partial m}, \frac{\partial r}{\partial m} = \frac{1}{9r}, \frac{\partial}{\partial t} \left(\frac{p^2 N}{2} \right) = 0 \text{ is set } v_{\ell} \left(r(t, n) \right) = \text{dis-}$

tance of the particles from the axis of the cylinder, g dentity, p - pressure) for the one-directed and undisturbed motion of an ideal gas with a constant thormal capacity mafter the

Card 1/3

On the Compression of a Gas Cylinder by Means of SUV/55-58-6-3/31 Current

introduction of the Lagrange variable m $(dm-p_1^{\circ}(R)RdR R=distance)$ tance of the particles from the cylinder axis at the time to and $g_1(R)$ - the initial distribution of density). The shock wave is taken into account by the following conditions: $g_1(g)=\xi$, $p=p_1^{\circ}+(1-\xi)g_1^{\circ}r^{\ast}$, $i-i_1^{\circ}=\frac{1}{2}(1-\xi^2)r^{\ast}$ (2) where g_1° , p_1° , i_1° denote the density, pressure and heat volume of the gas before the shock wave and g_1 , g_1° is the same behind the shock wave. If denotes the velocity of propagation of the shock wave. For the solution of the system of equations (1) the following series are set up in powers of ξ and inserted into (1) $g=g_0/\xi+g_1+\cdots$, $p=p_0+\xi p_1+\cdots$, $r=r_0+\xi r_1+\cdots$, Moreover, these expansions into series (3) and $r^*=r_0^*+\xi r_1^*+\cdots$ are introduced in the conditions for the shock wave (2). The series $r_0=r_0+\xi r_1+\cdots$ (c = cylinder) is set up for the motion of the external boundary of the cylinder and the pressure conditions are obtained (8). By means of the energy integral equation for the mentioned prob-

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On the Compression of a Gas Cylinder by Means of Current

sov/55-58-6-3/31

lem with the mentioned conditions, equations for the determination $r_0(t)$ and $r_{oc}(t)$ (9) and (11) and for $r^*(t)$ (10) may be obtained by an expansion into a series according to powers of $\mathcal E$. Further, a comparison is made of the results with those from reference 2. An agreement was found between the results of the compression periods of the cylinder and also with the experimental results. The method described here permits also the computation of all parameters of motion behind the shock wave. The numerical computations were carried out by means of an electronic digital computer. There are 1 figure and 4 Soviet references.

ASSOCIATION: Kafedra gidromekhaniki (Chair of Hydromechanics)

SUBMITTED: June 25, 1958

Card 3/3

SOV/179-59-1-32/36

AUTHOR: Lyubimov, G. A. (Moscow)

TITLE: Flow of Non-ideal Gas with a Great Supersonic Speed Around a Body (Obtekaniye tel potokom neideal'nogo gaza s bol'shimi sverkhzvukovymi skorostyami)

PERIODICAL: Izvestiya Akademii nauk SSSR, Otdeleniye tekhnicheskikh nauk, Mekhanika i mashinostroyeniye, 1959, Nr 1, pp 173-176 (USSR)

ABSTRACT: A method is described in Ref.l which is applied in this work to the case of supersonic flow round either a rotating body or a flat contour with an arbitrary relationship of pressure and temperature. The fundamental equations and their solutions are based on a system of coordinates as shown in Fig.l. When the function of the current \$\psi\$ is denoted as:

$$d\phi = \rho u r^{\nu - 1} dy - \rho v r^{\nu - 1} \left(1 + \frac{y}{R} \right) dx$$

then the equation of motion of gas in the system of coordinates Card 1/5

30V/179-59-1-32/36

Flow of Non-ideal Gas with a Great Supersonic Speed Around a Body

x, ϕ will take the form of Eqs.(1) where u and v - components of velocity along x and y; p, ρ , S - pressure, density, entropy of gas, R - radius of the contour of flow, γ - 1, 2 for a flat or rotating body respectively, r - distance from the axis of symmetry, α - angle between the tangent to the contour of the body and the direction of incoming flow. The solutions of the above equations in respect to the main shock wave are based on Eqs.(2), where γ - γ - parameters of gas in the incoming flow, i - heat content, γ - surface equation of the shock wave, γ - angle between tangent to the surface of the shock wave and the direction of incoming flow, γ - ratio of densities in flow and behind the shock wave which in the case of a curved wave becomes a function x . Generally, the solution of Eq.(1) can be obtained when the series:

$$p = p_0 + \varepsilon p_1 + \dots, \quad u = u_0 + \varepsilon u_1 + \dots, \quad \rho = \frac{\varrho_0}{\varepsilon} + \rho_1 + \dots,$$

$$v = \varepsilon v_0 + \dots, \quad y = \varepsilon y_0 + \dots,$$

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From this expression the value of $\varepsilon(x)$ can be determined.

$$i \left(\frac{2}{3} + 1 \right) a^{2} ais^{2} u = \left(\frac{0}{1} \varphi, \frac{0}{4} \right) i - \left(\frac{\frac{0q}{3}, 0q}{3} \right) i$$

where $y_0^* = y_0(x\phi_0^*)$ and the conditions to satisfy the equation $\phi = \phi^*$ are shown in Eqs.(4). These conditions are derived from the first three equations of Eqs.(2). The fourth equation of Eqs.(2) can be found in an identical way and the last equation of Eq.(2) can be expressed as:

$$\phi_{\mathbf{x}}(x) = \phi_{\mathbf{x}}^{\mathbf{y}} + \epsilon \phi_{\mathbf{x}}^{\mathbf{y}} + \cdots = \phi_{\mathbf{x}}^{\mathbf{y}} + \epsilon \phi_{\mathbf{x}}^{\mathbf{y}} + \cdots = \frac{\rho_{\mathbf{x}}^{\mathbf{y}} \sigma_{\mathbf{y}}}{\rho_{\mathbf{y}} \sigma_{\mathbf{y}}} + \sigma_{\mathbf{y}}^{\mathbf{x}} + \sigma_{\mathbf{y}$$

with a step ϵ are introduced. Thus the Eqs.(3) are derived. The limiting conditions will be defined when $\phi=\phi^*(x)$ which is an equation of shock wave in the coordinates x, ϕ . In this case:

Flow of Won-ideal Gas with a Great Supersonic Speed Around a Body

SOV/179-59-1-32/36

Flow of Non-ideal Gas with a Great Supersonic Speed Around a Body As an example, the problem of airflow around a cone is considered. In this case the flow of the shock wave is taken as linear and $\varepsilon = \text{const } (\mathbf{V} = 2)$. The solutions of Eqs.(3) and the limiting conditions of Eqs.(4) will take the form of the expressions to the left of Fig.4, p 175. The Figs.2 to 4 illustrate the results of the calculation when $\alpha = 25$, $T_1^0 = 220$, $p_1^0 = 0.1$ atm. The values of ε and T_0 are determined from:

$$\frac{\rho_1^{\circ}}{\epsilon} = \rho(T_0^{\circ}p_0), \quad i(T_0, p_0) - i(T_1^{\circ}, p_1^{\circ}) = \frac{1}{2}U^2 \sin^2 \alpha(1 + \epsilon) .$$

(ρ(T, p) and i(T, p) are taken from tables in Refs.4 and 6. The points in the figures marked with crosses denote the values when the characteristics of air at higher temperatures were considered. Fig.2 shows the relation of the density of the shock wave to the number M of the incoming flow. When M = 20 the temperature behind the shock wave rises to 3000°C, causing the dissociation of the air; the density of the shock wave increases more rapidly in comparison with that at lower Card 4/5 temperatures. Fig.3 gives the relation of the angle of the

30**V/179-**59-1-32/36

Flow of Non-ideal Gas with a Great Sipersonic Speed Around a Body shock wave β to the number M . Fig.4 gives the value of the coefficient of resistance as calculated from the formula at the bottom of p 175 (black dots represent the values taken from Ref.5). All the results were calculated from the theoretical formula and show a l to 1.5% accuracy in comparison with the same results obtained in the Laboratory of Physics of Fire in the Institute of Power, Academy of Sciences USSR, imeni G. M. Krzhizhanovskiy. There are 4 figures and 6 references, of which 2 are Soviet and 4 are English.

SUBMITTED: April 7, 1958.

Card 5/5

24 (3) AUTHORS:

Kulikovskiy, A. G. Lyubimov, G. A. SOV/179-59-4-16/40

(Moscow)

TITLE:

On the Possible Kinds of Crack With a Conductivity Jump

PERIODICAL:

Izvestiya Akademii nauk SBSR. Otdeleniye tekhnicheskikh nauk. Mekhanika i mashinostroyeniye, 1959, Nr 4, pp 130-131 (USSR)

ABSTRACT:

If in a flow of gas there is a surface with a jump-like change of its parameters, the mass-, momentum- and energy-conservation laws must be observed in the passing through this surface. Under certain assumptions made here, these laws are indicated in the form of formulas (1) (Ref 1). At given parameters of the approaching flow as well as of the electromagnetic field in front of the discontinuity surface, the formulas (1) determine the flow- and field parameters behind the discontinuity. It is shown that the presence of a single steady surface at given parameters of the approaching flow does not yet make it possible to solve only an unsteady problem with cracks of similar kind (e.g. the problem of the motion of a flat piston). The structure of the discontinuity surface with a conductivity jump is investigated. The procedure is similar to that described in the papers (Refs 2,3). The curve ABC shown in the figure is

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On the Possible Kinds of Crack With a Conductivity SOV/179-59-4-16/40 Jump

obtained. It expresses the connection between the volume v and the magnetic field strength H. It is shown that - if the structure of the discontinuity surface is investigated at $\delta = \delta(T)$, the conductivity δ being equal to zero, for T-values smaller than a certain T - there is only one point on the ABC-curve which depends on T*and the initial values of the parameters, and from which the motion can be continued until Δ . This points to a certain connection between H₁ and H₂,

which is not a consequence of the conservation laws, formula (1). This additional relationship, together with the conservation laws in unsteady problems, determines the intensity of the electromagnetic wave emitted, and makes the solution of such problems a unique one. There are 1 figure and 3 Soviet references.

SUBMITTED:

February 19, 1959

Card 2/2

67585 24.2120 SUV/179-59-5-3/41 Lyubimov, G.A. (Moscow) AUTHOR: Investigation of Stationary Surfaces of Discontinuity with TITLE: a Conductivity Step of a Gas in an Electromagnetic Field $\mathcal{V}^{ extsf{I}}$ PERIODICAL: Izvestiya Akademii nauk SSSR, Otdeleniye tekhnicheskikh nauk, Mekhanika i mashinostroyeniye, 1959. Nr 5. pp 9-15 (USSR) Surfaces of discontinuity in ordinary gas dynamics are ABSTRACT: well known, including their structure and stability. In magnetic gas dynamics surfaces of discontinuity have been studied in an infinitely conducting medium (eg Syrovatskiy, S.I. "On the Stability of Shock Waves in Magneto-Hydrodynamics" \ Zh ETF, 1958, Nr 6). The present work considers stationary surfaces of discontinuity wherein, apart from the thermodynamic quantities of flow and velocity, the conductivity of the gas also suffers a discontinuity. First, the effect of an electromagnetic field on discontinuities is considered generally and it is found that surfaces of discontinuity may appear in addition to those arising from ordinary gas dynamic laws. For example, a discontinuity combining a pressure rise and a drop in density is possible. The specific effect of changes in conductivity is then analysed. It is concluded Card 1/3

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SOV/179-59-5-3/41 Investigation of Stationary Surfaces of Discontinuity with a Conductivity Step of a Gas in an Electromagnetic Field

that not all surfaces of discontinuity which satisfy the laws of conservation can be considered as the limits of continuous flows with varying conductivity. Surfaces of discontinuity with a conductivity step which can be considered as limits of continuous flows are either density increases with a rise in the magnetic field or density reductions with a diminishing magnetic field. Taking into account the viscosity and heat conductivity of the gas, shows that stationary flows are only possible at certain initial values of the parameters. Thus, in the flow of an ideal gas containing a surface of discontinuity with a conductivity step, an electromagnetic wave will propagate ahead of the step which changes the initial parameters of the field. Finally, discontinuity surfaces with a conductivity step are considered on whose fronts energy is released. In the ordinary gas dynamics it is not possible to achieve detonation conditions wherein the velocity of the gas behind the detonation wave is larger than the velocity of sound unless energy is absorbed at the detonation front. If, however,

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67585 SOV/179-59-5-3/41

Investigation of Stationary Surfaces of Discontinuity with a Conductivity Step of a Gas in an Electromagnetic Field

the detonation proceeds in an electromagnetic field and the conductivity of the medium has a step at the detonation wave front, such conditions are possible at certain values of the initial parameters. Examples of such detonation waves were given by the author's earlier work ("Effect of Electromagnetic Fields on the Conditions of Detonation", DAN SSSR, 1959, Vol 126, Nr 3). There are 8 figures and 8 Soviet references.

SUBMITTED: March 13, 1959

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Card 3/3

In connection with V.A.Belokon's article "Permanent structure of shock waves with Joule dissipation." Zhur.eksp.i teor.fiz.
37 no.4:1173-1174 0 '59. (MIRA 13:5)

(Shock waves) (Belokon, V.A.)

10(6), 21(7)

AUTHOR:

Lyubimov, G. A.

SOV/20-126-2-18/64

TITLE:

A Shock Wave With a Discontinuity of the Conductivity of Gas in an Electromagnetic Field (Udamaya volna so skachkom provodimosti gaza v elektromagnituom pole)

PERIODICAL:

Doklady Akademii nauk SSSR, 1959, 701 126, Nr 2, pp 297-294 (7750)

ABSTRACT:

A gas current passing through a strong shock wave is bested in to such temperatures at which the gas is dissociated and ionize . Under these conditions and in the presence of an extential electromagnetic field, the gas current behind the shock wave must on no account be assumed to be non-conductive. Therefore, it is also of interest to investigate a shock wave with a discontinuity of gas conductivity on the front of this wave. The surhor here investigates a steady shock wave against which the current of a nonconductive gas (conductivity 64 =0) flows. Behild the shock wave the conductivity of the gas is assumed to be infinitely great $(\delta_n = \infty)$.

On the basis of these assumptions the relations on the shock wave (law of conservation for mass, momentum, energy, and for the tangential component of the electric field have the form

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A Shock Wave With a Discontinuity of the Conductivity of SOV/20-126-2-18/64 Gas in an Electromagnetic Field

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A Shock Wave With a Discontinuity of the Conductivity of SOV/20-126-2-18/54 Gas in an Electromagnetic Field

> wave, and the other quantities vary in the same manner as in an ordinary shock wave. The Hugoniot adiabatic of the shock wave under investigation has the equation

 $\xi_2 - \xi_1 + \frac{p_2 + p_4}{2} (v_2 - v_1) - \frac{1}{16\pi} (H_2^2 - H_1^2) (v_2 - v_1) - \frac{1}{4\pi} H_2 v_2 (H_1 - H_2) = 0$

where & denotes the internal energy of the unit of mass of the gas and V - the specific volume. With H1=H2 this adiabatic goes over

into the ordinary Hugoniot-adiabatic, and with E₁=0 it has the form $\xi_2 - \xi_1 + \frac{p_2 - p_1}{2} (v_2 - v_1) + \frac{H_1^2}{46\Pi} (v_1 + v_2) = 0$. Small disturbances with an infinite discontinuity of conductivity are impossible. Not only expansion discontinuities, but also compression discontinuities with a slight increase of pressure are forbidden. In the case of the permitted compression discontinuities higher degrees of compression are possible than in ordinary shock waves. With increasing pressure density behind these discontinuities decreases. As an example, the author investigates the steady flow of a gas current round plane contours, which takes place with supersonic velocity in an external

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A Shock Wave With a Discontinuity of the Conductivity of SOV/20-126-2-18/64 Gas in an Electromagnetic Field

electromagnetic field. In this case the conductivity of the gas behind the frontal shock wave becomes high at a very high velocity of the oncoming current, and within this range the equations of magnetic hydrodynamics must be applied. The problem of steadiness is not investigated. There are 2 figures.

ASSOCIATION:

Moskovskiy gosudarstvennyy universitet im. M. V. Lomonosova

(Moscow State University imeni M. V. Lomonosov)

PRESENTED:

February 11, 1959 by L. I. Sedov, Academician

SUBMITTED:

February 5, 1959

Card 4/4

10 (7) AUTHOR:

Lyubimov, G. A.

sov/20-126-3-20/69

TITLE:

The Influence of an Electromagnetic Field on the Development of a Detonation (Vliyaniye elektromagnitnogo polya na rezhim

detonatsii)

PERIODICAL:

Doklady Akademii nauk SSSR, 1959, Vol 126, Nr 3,

pp 532-533 (USSR)

ABSTRACT:

When solving the problem of the propagation of detonation waves of gases in electromagnetic fields, the fact must be taken into account that after the passage of the detonation wave the gas becomes electrically conductive. In the introduction to the present paper a steady detonation wave is investigated, and it is assumed that before the detonation wave the conductivity of the gas is equal to zero, and behind it, it is infinitely great. By basing on these assumptions, the conditions prevailing in the detonation wave are described by the system of equations (1), and for the adiabatic the equation (2) is given. In the first part of the paper a steady detonation wave without a field is first dealt with, and it is shown that in this case the aforementioned formula for the adiabatic goes over into that of the ordinary adiabatic. Next, the

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The Influence of an Electromagnetic Field on the Development of a Detonation

30V/20-126-3-20/69

process in the case of the existence of a field is investigated, and a formula is given for the propagation rate of minor disturbances behind the wave, from which it follows that their propagation rate behind the wave is higher, so that a decrease of density and pressure takes place. In the second part a cylindrical detonation wave propagating in a medium of constant density is investigated, and $\vec{E}_1 = -\frac{1}{c} \left[\vec{D} \vec{H}_1 \right]$ is assumed to hold. The system of equations (3) then describes the motion; as is shown by an analysis of this system, a decrease of pressure and density occurs, and it is impossible in ordinary cases to bring about this sort of detonation without loss of energy. There are 2 Soviet references.

ASSOCIATION: Moskovskiy gosudarstvennyy universitet im. M. V. Lomonosova

(Moscow State University imeni M. V. Lomonosov)

PRESENTED:

February 11, 1959, by L. I. Sedov, Academician

SUBMITTED:

February 5, 1959

Card 2/2

. 10(4)

AUTHOR:

Lyubimov, G. A.

TITLE:

The Steady Flow Round a Corner of the Current of an Infinit ly Conductive Gas (Statsionarnoye obtekaniye ugla potokom besko-

SCV/20-126-4-12/62

nechno provodyashchego gaza)

PERIODICAL:

Doklady Akademii nauk SSSR, 1959, Vol 126, Nr 4,

pp 733 - 735 (USSR)

ABSTRACT:

By means of the system of equations (1), the system of equations of magnetic hydrodynamics in the case of an infinitely great conductivity of the medium is represented in polar coordinates, and in the first part, the solutions of (1) are given for a progressive stream and a rotational wave according to Prandtl-Mayer. For the flow round an infinitely conductive angle there are solutions which depend only on the coordinate $\mathfrak{p};$ the external magnetic field must be parallel to the surface of the angle. These solutions are developed and written down with the system of equations (6). For determination of the function $\mathfrak{q}(\mathfrak{p})$ the ordinary differential equation (7) is integrated and then $\mathfrak{q}(\mathfrak{p})$ is substituted into (6). Finally, the flow round of a non-conductive angle is dealt with. For the solution of this problem, the general system of equations (1) must be calculated and with equation (8) the integral of this system is written

Card 1/2

The Steady Flow Round a Corner of the Current of an SOV/20-126-4-12/62 Infinitely Conductive Gas

down. The results obtained show that, if an infinitely conductive gas flows round an angle, the velocity v is equal to the velocity of sound, similar to what is the case in ordinary gas dynamics. There are 1 figure and 2 Soviet references.

ASSOCIATION: Moskovskiy gosudarstvennyy universitet im. M. V. Lomonosova

(Moscow State Jniversity imeni M. V. Lomonosov)

PRESENTED: February 21, 1959, by L. I. Sedov, Academician

SUBMITTED: December 3, 1958

Card 2/2

66467

2:(7) 10.2000(A)

SCV/20-129-1-15 64

AUTHORS:

Kulikovskiy, A. G., Lyubimov, G. A.

TITLE:

Magnetchydrodynamic Gas-ionizing Shock Waves

PERICHICAL: Doklady Akademii nauk SSSR, 1959, Vcl 129, Nr 1.

pp 52 55 (USSR)

ABSTRACT:

An electromagnetic wave may move in front of a shock wave in unsteady problems, in which shock waves ionize the das. present in an electromagnetic field. For known velocity of the gas behind the shock wave, the boundary conditions in the shock wave (expressing the continuity of the tangential component of the electric field as well as the fluxes of matter, momentum. and energy) are not sufficient to determine simultaneously the intensities of the shock wave and of the emitted electromagnetic wave An additional relation between quantities before and to hind the shock wave is furnished by the investigation of the structure of the shock waves of the above type. This rolls and, in consequence, the alteration of all quantities in the shock wave depends essentially on the amount of the relations between the dissipation coefficients (viscosity, thermal on ductivity, and magnetic viscosity) in the transition with

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Magnetony drody namic Gas-ionizing Shock Waves

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The electrical conductivity of the gases is considered in a function of temperature in the present paper: $Set(T) = I^{1/2} + I^{1/2}$ that $\sigma=0$ if T(T and $\sigma>0$ for T>T% . The structure if a cuting dynamic shock wave, which moves in a gas at a temperature TAX was investigated by the authors. For simplicity only cases are treated, in which only 2 dissipation coefficients that are magnetic viscosity and molecular viscosity or magneti viscosity and thermal conductivity) are not equal : electric and the magnetic field are assumed to the entropy was to each other and in parallel to the plane of the wave frint The rather extensive equations of the magnetonyar dynamic are written fown for both cases and shortly explained. Trese differe ential equations fix the family of integral curves of the plane H, v (where it holds that $\sigma > 0$, $\gamma_m \neq \infty$) aH= const in the range TK THE shock wave may be represented by solutions of all t kini, which pass over into a progressive flow if $x = \pm \infty$. If r these solutions all derivations converge towards as x approaches +00 First, a gas, which moves from x=-00, is subject . Pasty natical compression and at Ty TW the gas starts to prices to

Carl 2/4

4- -- 7 Magnet thy intriguance Gastion; sing Shook Waves = S0V/20.139-1 14 €3 with the magnetic field. The change of the magnetic field in the wave Ho . H. is determined by the point of interpretion of the integral curve and the line $T_{\theta}T^{\frac{1}{2}}$. This joint of intersection depends on the characteristics of the incoming to work well as on the ratio of the dissipation coefficients within the transition zone. The relation H. H. (B. ... T. .v. y. -lic an additional boundary condition on the substitution of a snock wave for a steady flow. If one of the dissipation overfitcients is considerably greater than the others, this admitters. boundary condition may be ascertained in explicit form. The width of the shock waves is defined by the greatest in f the dissipation poefficients. There are fifther and product ences. 3 of which are Soviet ASSOCIATION: Matematicneskly institut im V A Steklova Akalemii hask CDUR (Mathematical Institute ideni V. A. Steklov of the A utbuy of Sciences, USSR) Cari 3/4

Magnetonydrudynamic Gas-ichizing Shick Waves SCV, 20.123.1119 82
PRESENTEL: June 30. 1959, by L I Sedov, Academician
SUBMITTEL: June 20. 1959

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21-(7), 24-(3) 24.2/20, 10.2000(A)
AUTHORS: Kulikovskiy, A. G., Lyubimov, G. A. 66448 SOV/20-1 29-3-14/70

The Simplest Problems Concerning a Gas-ionizing Shock Wave in TITLL:

an Electromagnetic Field

Doklady Akademii nauk SSSR, 1959, Vol 129, Nr 3, pp 525-528 (USSR) PERIODICAL:

If the conductivity of the gas before the shock wave vanishes and is finite behind the shock wave, the theorems of ABSTRACT:

conservation read: $g_1 v_1 = g_2 v_2$, $p_1 + g_1 v_1^2 + (1/8\pi)H_1^2 =$

= $p_2 + 9_2 v_2^2 + (1/8\pi)H_2^2$, $9_1 v_1 \left(\frac{v_1^2}{2} + i_1\right) + (c/4\pi)E_1 H_1 =$

= $g_2 v_2 \left(\frac{v_2^2}{2} + i_2 \right) + (c/4\pi) E_2 H_2$, $E_1 = E_2 = \frac{v_2}{c} H_2$. The electric and the magnetic field strength are, for the purpose of

simplifying matters, assumed to be parallel to the wave front and perpendicular to each other. The shock waves ionizing a gas may be considered to be the limit of a certain continuous motion of a viscous heat-conducting gas,

the conductivity 6 of which is considered to be a known

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The Simplest Problems Concerning a Gas-ionizing Shock Wave in an Electromagnetic Field

function of the temperature T (T < T*, σ > 0 at T > T*). This as well as other facts mentioned here indicate the following: The solution of problems concerning ionizing shock waves will differ from the solutions of the corresponding problems in gasdynamics and magnetogasdynamics. This difference exists not only in the electromagnetic wave, but also in the variation of the gas-dynamical parameters of the motion. In gasionizing shock waves compression is not higher than in gasdynamic shock waves and not less than in magnetogasdynamic shock waves which have the same parameters of the incoming flow and the same magnetic field strength before the discontinuity. Also the other quantities behind the gasionizing shock wave attain values which are between the corresponding values behind the gas-dynamic shock wave and a magnetogasdynamic shock wave. in the first part of the present paper the problem of the motion of a plane piston is dealt with. In this case the presence of an electromagnetic field increases the velocity of the shock wave and reduces the compression in it compared to the gasdynamic solution at the same piston velocity. The second part deals with the flow

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The Simplest Problems Concerning a Gas-ionizing Shock Wave in an Electromagnetic Field

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round a wedge. The velocity component which is tangential with respect to the shock wave remains conserved during passage through the shock wave, and the variations of normal velocity and of the other quantities may be dealt with in the same manner as in the first part. A surface charge must exist on the shock wave. There are 2 figures and 3 Soviet references.

PRESENTED:

July 14, 1959, by L. I. Sedov, Academician

SUBMITTED:

July 7, 1959

4

Card 3/3

"On Gas-Ionizing Magnetohydrodynomic Thock Daves."

report presented at the Intl Cymposium on Magneto-Pluid Dynamics, 17-24 Jun 1971, Wash., Comments - P 3,151,565, Oh Feb Co.

Testing the elastic materials of a turbodrill shoe under static and dynamic load conditions. Izv. vys. ucheb. zav.; meft' i and dynamic load conditions. Izv. vys. ucheb. zav.; meft' i (MIEA 14:1) gaz 3 no.11:25-32 '60. 1. Moskovskiy institut neftkhimicheskoy i gazovoy promyshlennosti imeni akademika I.M. Gubkina i Vessoyuznyy nauchno-isəledovatel'skiy imeni akademika institut burovoy tekhniki. (Turbodrills—Testing)

10.8000

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AUTHORS:

TITLE:

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89394 5/040/61/025/001/013/022 B125/B204 Kulikovskiy, A. G., Lyubimov, G. A. (Moscow) The structure of an inclined magnetohydrodynamic shock wave

Prikladnaya matematika i mekhanika, v. 25, no. 1, 1961, PERIODICAL:

TEXT: The present paper investigates the flow within the zone of the shock wave when the dissipation of energy in the wave is caused by magnetic viscosity and by the second kinematic viscosity. In the problem of the structure of a magnetohydrodynamic shock wave, the solutions of the equations of the magnetohydrodynamics of a non-perfect gas are to be determined, whose values with x = + 00 satisfy the known laws of conservation. If only the magnetic viscosity and the second viscosity are nonvanishing, the equations of the steady one dimensional flows of a perfect gas read $v_m \frac{dH}{dx} = uH - vH_n + cE, \quad \mu \frac{du}{dx} = p + \rho u^2 + \frac{1}{8\pi}H^3 - J_1$

 $\rho uv - \frac{1}{4\pi} H_n H = J_2, \qquad \rho u = M, \qquad H_n = \text{const}$ $\rho u \left[\frac{\gamma}{\gamma - 1} \frac{p}{\rho} + \frac{1}{2} (u^2 + v^1) \right] - \frac{cEH}{4\pi} = U$

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They refer to a system of coordinates, in which the flow is plane. Hn, H, u, v are the components of the magnetic field and of the velocity along the x- and y-axis, E - the z-component of the electric field, c - velocity of light; J₁,J₂ - the fluxes of the x- and y-components of the momentum, U - the energy flux, M - the mass flux. With the dimensionless variables $u = u_0 \tau$, $v = u_0 q$, $p = Q_0 u_0^2 \theta$, $H = \sqrt{4\pi Q_0 u_0^2 h}$ (2) one obtains $\frac{v_m}{u_0} \frac{dh}{dx} = h(\tau - h_n^2) - e, \qquad \frac{\mu}{\rho_0 u_0} \frac{d\tau}{dx} = \theta + \tau + \frac{1}{4}h^2 - P$

$$q - h_n h = 0, k0\tau + \frac{1}{2}\tau^2 + \frac{1}{3}h_n^3h^3 + eh = \varepsilon$$

$$\frac{H_n}{1, h_n} = \frac{H_n}{\sqrt{\frac{1}{2}\pi\rho_0 u_0^3}}, e = -\frac{eB}{\sqrt{\frac{1}{2}\pi\rho_0 u_0^3}}, P = \frac{J_1}{\rho_0 u_0^3}, \varepsilon = \frac{U}{\rho_0 u_0^3}$$

 $\left(k = \frac{\gamma}{\gamma - 1}, \ h_n = \frac{H_n}{\sqrt{2\pi\rho_0 u_0^2}}, \ e = -\frac{cE}{\sqrt{4\pi\rho_0 u_0^4}}, \ P = \frac{J_1}{\rho_0 u_0^3}, \ \epsilon = \frac{U}{\rho_0 u_0^3}\right),$ Furthermore, $e = h_0(1 - h_n^2)$, $P = 1 + \theta_0 + \frac{1}{2}h_0^2$, $\epsilon = k\theta_0 + \frac{1}{2} + h_0^2(1 - \frac{1}{2}h_n^2)$ holds. Besides, everywhere e>0 is assumed. For reasons of simplicity, here $\gamma < 2$ is assumed. The real points of the isoclinal line $d\tau/dx = 0$

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The structure of an inclined ... are on both sides of the hyperbola $h = e/(k\tau - h_n^2)$. The maxima and minima of the isoclinal line are on this hyperbola on such points τ , where the discriminant of $h^2(k\tau-h_n^2)$ - 2eh + $(2k-1)\tau^2$ - $2kP\tau$ + 2ϵ = 0 (6) is equal to zero. The isoclinal line $d\tau/dx=0$ has the asymptote $\tau=h_n^2/k$. With increasing α the roots of $D(\tau) = 2\alpha(P-1)\left[(1-kh_n^2)+k(h_n^2+k-2)\tau\right]-(k\tau-h_n^2)\left[(2k-1)\tau^2+k(h_n^2+k-2)\tau\right]$ $-2kP\tau + 2k(P-1)+1$ change monotonically. In the plane of the variables $P - 1 = \frac{1}{2}h_0^2 + \theta_0$ and h_n^2 , there is a curve which separates the domain of existence of the three roots of the discriminant from that of a single root with $\theta_0 = 0$ ($\alpha = 1$) (see Fig. 1, curve ABCD). The curve ECF illustrating the equation $\tau = \tau_{*}$ touches the curve ABCD at the point C. To the left of ABCF, the discriminant has three roots with small α , and with large α it has one root. For the remaining points of the variable $P-1,h_{n}^{2}$, the discriminant, with small and large α , has three roots, but with intermediary values of α , it has one single root. Case a): In the Card 3/7

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The structure of an inclined ...

case of points lying simultaneously below the straight line $\tau_1 = h_2^2/k$ and $h_n^2/k = 1$, two roots of the discriminant are greater than h_n^2/k , and one is smaller than h_n^2/k . Case b): In all other cases with three roots, one root is greater than h_n^2/k , and the two others are smaller. These properties permit the construction of the isoclinal line. For points above the straight line $\tau = h_n^2$, the velocity is greater than Alfvèn velocity $a_{A} = H_{n} / \sqrt{4\pi Q_{0}}$, and for points below this straight line it is smaller than Alfven velocity. To the states before and behind the shock wave there correspond the points of intersection of the isoclinal lines (6) and (8). To the solution of the problem of the structure of the shock wave, there corresponds the integral curve of the Eq. (9)

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points lying in the region $\tau > 0$. With continually decreasing velocity, the following singular points are possible: 1) Nodes, 2) saddle, 3) saddle, 4) nodes, into which the integral curves lead. If the curves (6) $h^2(k\tau-h_n^2)$ -2eh+(2k-1) τ^2 -2kP τ +2 ϵ = 0 have the shape indicated in Figs.2 and 4, then all singular points lie on the same branch of the curve (6). In Figs. 5 and $7 \mu \varrho_0 \nu_m$ is either small or large, respectively. In Fig.6, the single value of $\mu/\varrho_0 v_m$, at which the integral curve emerging from point 2 runs into point 3, corresponds to the value of $(\mu/\varrho_0\gamma_m)_*$. the slow waves thus have a structure with an arbitrary ratio of dissipative coefficients. In four singular points the structure may also have intermediary shock waves. The transition $2 \rightarrow 3$ is possible only in the case of $\frac{\omega}{2}$, the transitions 1-+3 and 2-+4 exist and are unique with $\left(\frac{\omega}{Q_0 V_m}\right)_*$, and the transition 1->4 is possible with $\frac{\omega}{Q_0 V_m} > \left(\frac{\omega}{Q_0 V_m}\right)_*$,

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and may also occur on an infinite number of integral curves. The structure of the "evolution" shock waves (in the sense of A. I. Akhiyezer et al.) differs from the structure of the non-evolution shock waves by the fact that only they have a structure at any ratio between the dissipative coefficients. A. N. Voynov is mentioned. There are 7 figures and 4 references: 2 Soviet-bloc and 3 non-Soviet-bloc.

SUBMITTED: July

July 16, 1960

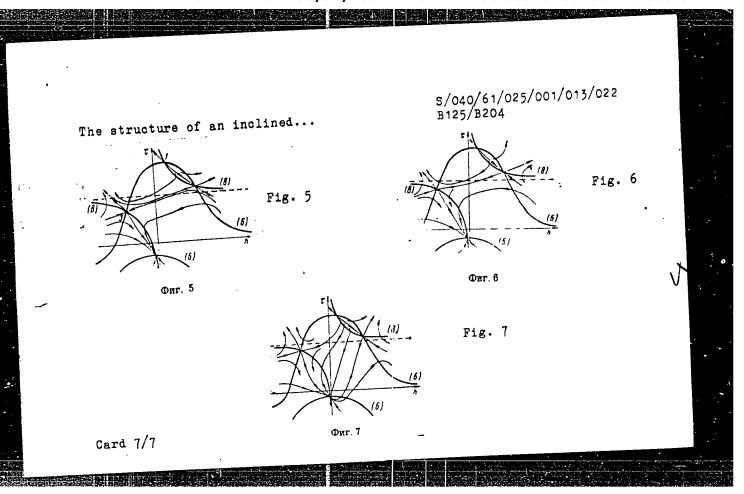
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Fig. 1

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AUTHOR:

Dyubimmov, G.A. (Moscow)

TITLE:

The structure of magneto fluid dynamic shock waven

in a gas with anisotropic conductivity

PERIODICAL: Prikladnaya matematika i mekhanika, v. 25, no. 2,

1961, 179 - 186

TEXT: For a gas with anisotropic conductivity flowing in a magnetic field which must be wide enough to allow for a suital movement of electrons the following will apply:

$$\omega \tau = \frac{eV}{m_e c}$$

where ω - harmor's frequency; τ - time between collisions of electrons and ions. If Eq. (1) is to hold true for the ionization of gas, Ohm's law must be satisfied according to T. Kautling (Ref. 1: Magnitnaya gidrodinamika, Izd. inostr. lit., 1959).

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The structure of magneto ...

$$\sigma(E + \frac{1}{c} v \cdot H + \frac{1}{ne} \operatorname{grad} p_e) = j + \frac{\omega \tau}{H} j \cdot H$$
 (2)

where o - conductivity in the absence of magnetic field; p_c - cross of electrons; n - number of electrons in unit volume; c - electrons charge. Hence one can assume hydrodynamic conditions within the gas. To investigate the structure of the shock wave or embauses that it occupies a narrow zone in the current field. In this zone, the parameters of the current change slowly. As the wave dies away the current becomes steady. It is assumed that in the core of the wave, only dissipation energy of electric current is of any importance. The x-axis is taken at right angles to the normal of a wave's surface and the y and z lie in the wave's plane. Then, for the core of the wave (Maxwell's equations being satisfied) the dimensionless condition

fied) the dimensionless condition
$$u = u_0 u^*, \quad v = u_0 v^*, \quad w = u_0 w^*, \quad RT = \theta u_0^2, \quad u^*p = \rho_0 u_0^2 \theta \quad (5)$$

 $H_{i} = \sqrt{8\pi \gamma_{0} u_{0}^{2} h_{i}}$ (i = x, y, z)

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The structure of magneto ...

holds, where 0 - temperature, h_i - the dimensionless component of the vector of magnetic field intensity, u*, v*, w* - dimensionless components of velocity [Abstractor's note: Other symbols not defined]. Before the shock-wave

$$w_0 = H_{Z0} = J_3 = 0$$
 (6)

holds. [Abstractor's note: $J = J_1$, J_2 , J_3 , not defined]. Simplifying and eliminating gives

$$\frac{1}{2} (\gamma + 1) u^{*2} + \gamma u^{*} (h_{y}^{2} + h_{z}^{2}) + 2 (\gamma - 1) \epsilon_{x}^{2} (h_{y}^{2} + h_{z}^{2}) + 2 (\gamma - 1) J_{z}^{*} h_{x} h_{y} + \frac{1}{2} (\gamma - 1) = (\gamma + 1) \varepsilon^{*} = 0$$
(10)

where asterisks denote dimensionless coordinates, and the vectors U and H are known to be parallel before and after the wave. Eq. (10) determines the structure of the slock wave. If $h_z = 0$, (10) gives the curve for the structure of a magneto-hydrodynamic shock-card 3/9

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The structure of magnete ...

wave in an isotropic gas. If only one dissipation coefficient of is given then

$$\frac{1}{2}(\gamma + 1)u^{*2} + \gamma u^{*}h_{y}^{2} + 2\gamma_{1} + 4\gamma_{1}h_{y}^{2}h_{z}^{2} + \gamma J_{1}^{*}u^{*} - 2(\gamma + 1)J_{2}^{*}h_{x}h_{y} + \frac{1}{2}(\gamma + 1)J_{z}^{*2} + (\gamma + 1)e^{*} = 0$$
(11)

The surface expressed by Eq. (10) is found along the perpendicular axis u* whose center lie on the hyperbola

$$h_{y} = \frac{(\gamma - 1) J_{2}^{*}}{\gamma u^{*} - 2(\gamma - 1) \gamma_{x}^{2}}.$$
 (12)

In the case
$$x = \pm \infty$$
,
 $(u^* - 2h_x^2)(\alpha^*h_x^h_y + h_z) - \alpha^*h_x^2J_2^* = 0$ (13)

$$(u^* - 2h_x^2)(L_y - \alpha^*h_xh_z) - h_x J_2^* = 0$$
 (14)

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The structure of magneto ...

From the above statements one may determine that in the vicinity of u^* , h_y , h_z , corresponding to conditions before and after the wave, must lie intersections of the surface expressed in (10) and hyperbolic cylinders Eq. (13) and Eq. (14). Also Eqs. (13) and (14) lie on the hyperbola $(u^* - 2h_x^2) h_y - h_x^{J_2}^* = 0$ (15)

which lies in the plane h = 0. Therefore it is seen that a point after and before the wave lies in the plane h = 0 and occurs at the points of intersection of the curves (11) and (15). The intersections of curves (11) and (15) depend on parameters before and after the shockwave. Investigating local properties of intersection points of the curves (11) and (15) one assumes that u* = 1, tion points of the curves (11) and (15) one assumes that u* = 1, y = h, h = 0. This may be obtained by scaling down coordinates of the axis. Individual points of intersection may be distinguished by considering the discriminant D,

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The structure of magneto ...

$$D = \left[-\frac{h_{y_0}^{3}}{1 - \gamma \theta_0} + (1 - 2h_x^{2}) \right]^{2} + (1 - 2h_x^{2}) \left[\frac{2h_{y_0}^{3}}{1 - \gamma \theta_0} - (1 - 2h_x^{2}) \right] \times$$

$$\times (1 + \alpha^{*2}h_x^{3}) = \frac{h_{y_0}^{4}}{(1 - \gamma \theta_0)^{3}} + \alpha^{*3}h_x^{2} (1 - 2h_x^{2}) h_{y_0} \left[\frac{2h_{y_0}}{1 - \gamma \theta_0} - \frac{1 - 2h_x^{3}}{h_{y_0}} \right]$$
(23)

If D > 0, then the point is a node if

$$\frac{2h_{yo}}{1-\gamma\theta_{o}} > \frac{1-2h_{x}^{2}}{h_{yo}}$$
 (21)

is satisfied, and a saddle-point if

$$\frac{2h_{yo}}{1-\gamma\theta_{o}} = \frac{1-2h_{x}^{2}}{h_{yo}}.$$
 (20)

If D < 0 the point is a focus. In the case of rapid waves all nodes and foci are initial, and in the case of slow waves the integrated card 6/9

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The structure of magneto ...

gral curves always pass through nodes and foci. The author concludes that the current in the internal zone depends primarily on the spiral flow of electrons and on the wave it all (fast or slow). Inside the wave, vectors of intensity of a magnetic field may change over if there is a large spiral flow of electrons. This movement does not follow the rotation of vector H around one x-axis. If the spiral flow of electrons tends to zero ($\alpha^* \rightarrow 0$) then the shock wave changes to an ordinary magneto-fluid dynamic shockwave. Beyond the amplitude of the shockwave, if it is strong, there may exist a region where magnetic field is the same as in the front of the shock wave. If this is so, with $\omega \tau \sim 1$, becomes

$$l \sim \frac{c^2}{4\pi\sigma u_0} (1 + \alpha^2 H_x^2) \sim \frac{c}{4\pi\sigma u_0} (1 + \omega^2 \tau^2)$$

or with wt ≫1

$$l \sim \frac{c^2}{4\pi \sigma u_0} \frac{1 + \alpha^2 H_x^2}{\alpha H_x} \sim \frac{c^2}{4\pi \sigma u_0} \omega \tau$$

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The structure of magneto ...

Hence the amplitude of the magneto-fluid-dynamic shockwave in an isotropic gas is greater than an ordinary magneto-fluid-dynamic wave, and, therefore, for large spiral flow of electrons (ωt - 1), the amplitude of the shockwave in an isotropic gas conforms with Larmor's radius for ions

$$l \sim \frac{c^2}{4\pi\omega_0} \omega \tau = \frac{cH}{4\pi\omega_0 nl} \sim \frac{cH^2\omega_0 m_i}{4\pi\rho\omega_0^2 lH} \sim R_i$$

In this case energy of the magnetic field is equal to the mean vinematic energy $(H^2/8\pi \sim u_0^2)$. Projecting Eq. (2) on the x-axis gives

 $-\frac{\omega \tau}{H}\frac{d}{dx}(H_v^2 + H_z^2) = \frac{\sigma}{c}(vH_z - wH_v) + \frac{1}{ne}\frac{dp_e}{dx} + \sigma E_x \qquad (24)$

This equality can be used in determining $E_{\rm X}({\rm x})$ when ${\rm p_e}$ is known. If the random velocities of ions and electrons are of the same order and are expressed in whole numbers, then 2 ${\rm p_e}$ = p (${\rm p_l}$ = ${\rm p_e}$). Card 8/9

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The structure of magneto ...

With $E_{\mathbf{x}}(\mathbf{x})$ known, equality $dE_{\mathbf{x}}/d\mathbf{x}=4\pi p_{\mathbf{e}}$ determines the density of charge inside the wave as a function of \mathbf{x} . There are 4 figures and 6 references: 5 Soviet-bloc and 1 non-Soviet-bloc.

SUBMITTED: July 16, 1960

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2673h S/040/61/025/003/011/026 D208/D304

AUTHORS: Baranov, V.B., and Lyubimov, G.A. (Moscow)

TITLE: Generalized Ohm's law in a completely ionized gas

PERIODICAL: Akademiya nauk SSR. Otdeleniye tekhnicheskikh nauk. Prikladnaya matematika i mekhanika, v. 25, no. 3,

1961, 468 - 472

TEXT: In deriving equations of motion of fully ionized gas and relations connecting current density with other parameters, the concept of a binary (electron-ion) mixture is used. Here the problem considered is that of the influence of viscosity of the components on the equation for the curr t density of generalized Ohm's law and the dimensionless criteria are given which influence the final form of the generalized Ohm's law for a completely ionized gas. The gas is assumed to consist of the electrons and singly charged ions and their number per unit volume to be n. The equations of motion for each component are

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Generalized Ohm's law in a ...

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$$m_{e}n\frac{d_{e}\mathbf{v}_{e}}{dt} = -\nabla p_{e} - \operatorname{div}\boldsymbol{\pi}_{e} + cn\left(\mathbf{E} + \frac{1}{2}\left\{\mathbf{v}_{e} \wedge \mathbf{H}\right\}\right) + \mathbf{R}_{e}$$
 (1)

$$m_{i}n\frac{d_{i}\mathbf{v}_{i}}{dt} := -i\nabla p_{i} - \operatorname{div}\boldsymbol{\pi}_{i} + en\left(\mathbf{E}_{i} + \frac{1}{c}\left\{\mathbf{v}_{i} \times \mathbf{H}\right\}\right) + \mathbf{R}_{i}$$

$$\frac{d_{e}}{dt} = -\frac{\partial}{\partial t} + \mathbf{v}_{e}\nabla_{i}, \quad \frac{\partial}{\partial t} = -\frac{\partial}{\partial t} + \mathbf{v}_{i}\nabla$$
(2)

where m_e , m_i = mass of the electron and iron respectively (m_e m_i) v_e , v_i = macroscopic velocities; p_e , p_i = partial pressures; e_e , e_i = tensors of viscous stresses for the electron and ion gas respectively, e_i = electron charge, E_i and E_i = intensities of electric and magnetic fields, and E_i = E_i . If also E_i = E_i

$$\frac{en\left(-\frac{d_{i}\mathbf{v}_{e}}{dt} + \frac{di_{i}\mathbf{v}_{i}}{dt}\right) - \frac{e}{m_{e}}\nabla p_{e} + \operatorname{div}\left(\frac{e}{m_{e}}\pi_{e} - \frac{e}{m_{i}}\pi_{i}\right) + \frac{e^{2}n}{m_{e}}\left(\mathbf{E} - \frac{1}{e}\mathbf{v} + \mathbf{H}\right) - \frac{e}{m_{e}}\mathbf{j} + \mathbf{H} - \frac{e^{2}n}{n_{e}}\mathbf{j} + \frac{1}{3}\mathbf{j} - (m_{e} < m_{i})}$$
(3)

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Generalized Ohm's law in a ...

$$\mathbf{j} = -\frac{ne^{2}\mathbf{r}_{e}}{m_{e}}, \qquad \mathbf{j} = -en(\mathbf{v}_{e} - \mathbf{v}_{i})$$

where o = conductivity of the gas in the absence of magnetic field. j = current density, τ_e - time between two electron-ion colli-

sions. If in addition $m_i v_i = m_e v_e$ and $v = v_i$, $v_e = v - \frac{1}{en} j$, then together with the continuing $\frac{dn}{dt} + n$ div v = 0

together with the continuing
$$\frac{1}{dt} + n \frac{dv}{dt} + \frac{e^2n}{m_e} (E + \frac{1}{c} v / H) - \frac{dj}{dt} + j \frac{div}{dt} v + (j \overline{v})v - (j \overline{v}) \frac{j}{en} = \frac{e^2n}{m_e} \frac{1}{o} j + \frac{e^2n}{m_e} (E + \frac{1}{c} v / H) - \frac{dj}{dt} + j \frac{div}{dt} v + (j \overline{v})v - (j \overline{v}) \frac{j}{en} = \frac{e^2n}{m_e} \frac{1}{o} j + \frac{e^2n}{m_e} (E + \frac{1}{c} v / H) - \frac{dj}{dt} + j \frac{div}{dt} v + (j \overline{v})v - (j \overline{v}) \frac{j}{en} = \frac{e^2n}{m_e} \frac{1}{o} j + \frac{e^2n}{m_e} (E + \frac{1}{c} v / H) - \frac{dj}{dt} + j \frac{div}{dt} v + (j \overline{v})v - (j \overline{v})v - (j \overline{v})v + (j \overline{v})v - (j \overline{v})v - (j \overline{v})v + (j \overline{v})v - (j$$

$$-\frac{e}{m_e c} j \times H + \frac{e}{m_e} \nabla p_e + \text{div} \left(\frac{e}{m_e} \mathcal{L}_e - \frac{e}{m_i} \mathcal{L}_i\right)$$

is obtained. It is assumed that characteristic time the max $\{\tau_e, \sigma_e\}$

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Generalized Ohm's law in a ...

 τ_{i} . When the electromagnetic field influences the motion and viscous forces are present,

forces are present,
$$\int V^2 = \frac{1}{c} jHL, \text{ or } j = \frac{nm_i V^2 c}{HL} \qquad (\zeta = n(m_e + m_i) = nm_i) \qquad (5)$$

and

$$\eta \frac{V}{L} = V^2 \text{ or } \eta = 0.96 \text{nTr}_1$$
 (6)

where V, L = characteristic velocity and Length associated with the problem. T = temperature. From Eqs. (5), (6) the following expressions are obtained for the terms of (4).

$$A_1 = \mathbf{j} \operatorname{div} \mathbf{v} \sim (\mathbf{j} \bigtriangledown) \mathbf{v} \sim \frac{enm_1 V^2 eV}{eH L^4} = enV \frac{\Omega^2}{\omega_1}$$

$$A_2 = (\mathbf{j} \bigtriangledown) \frac{\mathbf{j}}{en} \sim \frac{n m_i V^2 e n m_i V^2 e}{H L e n L H L} = e n V \frac{\Omega^3}{\omega_{i2}}$$

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CIA-RDP86-00513R001031210008-5" **APPROVED FOR RELEASE: 08/31/2001**

Generalized Ohm's law in a ... S/040/61/025/003/011/026 $A_{3} = \frac{e^{2n}}{m_{e}} \frac{1}{1} \mathbf{j} = \frac{1}{t_{e}} - \frac{mn_{e}^{1/2}}{m_{e}^{2}} - enV \frac{\Omega}{\omega_{e}^{2}t_{e}}$ $A_{4} = \frac{e^{2n}}{m_{e}} \mathbf{E} \geq \frac{e^{2n}}{m_{e}^{2}} \|\mathbf{v} - \mathbf{H}\|_{2} - enV \omega_{e}$ $A_{5} = \frac{e}{m_{e}} \nabla p_{e} \leq \frac{e}{m_{e}^{2}} \mathbf{j} - \mathbf{H} = \frac{enV}{m_{e}^{2}} \frac{m_{e}^{2}}{m_{e}^{2}} = 0.96 \text{ nT } \mathbf{T} \mathbf{I} \frac{\tau_{e}}{m_{e}^{2}} \frac{v_{e}^{2}}{\tau_{e}^{2}} = 0.96 \text{ nT } \mathbf{T} \mathbf{I} \frac{\tau_{e}^{2}}{m_{e}^{2}} = 0.96 \text{ nT } \mathbf{T} \mathbf{I} \frac{\tau_{e}^{2}}{m_{e}^{2}} = 0.96 \text{ nT } \mathbf{T} \mathbf{I} \frac{\tau_{e}^{2}}{m_{e}^{2}} = 0.96 \text{ nT } \mathbf{I} \frac{\tau_{e}^{2}}{m_{e}^{2}} = 0.96 \text{ nT } \frac{\tau_{e}^{2}}{m_{e}$

26734 S/040/61/025/003/011/026 D208/D304

Generalized Ohm's law in a ...

where Ω = characteristic frequency and all the terms of Eq. (4) are expressed in dimensionless parameters Ω/ω_i and $\omega_e \tau_e$. The final form of Ohm's law will depend on those parameters, and the following cases are considered: 1) $1/\omega_i$ 1, $\omega_e \tau_e$ 1; 2) $1/\omega_i$ 1, $1/\omega_e \tau_e$ 1; 3) $1/\omega_i$ 1, $1/\omega_e \tau_e$ 1; 4) $1/\omega_i$ 1, $1/\omega_e \tau_e$ 1; 5) $1/\omega_i$ 1; $1/\omega_e \tau_e$ 1; 6) $1/\omega_i$ 1, $1/\omega_e \tau_e$ 1; 7) $1/\omega_i$ 1, $1/\omega_e \tau_e$ 1; 8) $1/\omega_i$ 1, $1/\omega_e \tau_e$ 1; 9) $1/\omega_i$ 1, $1/\omega_e \tau_e$ 1. The result shows that in deriving Ohm's law for a binary model of a completely ionized gas, the viscosity terms can be neglected. There are 3 Soviet-bloc references.

SUBMITTED: March 4, 1961

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"APPROVED FOR RELEASE: 08/31/2001 CIA-RD

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26124

S/040/61/025/004/003/021 D274/D306

26.731

AUTHOR:

Lyubimov, G.A. (Moscow)

TITLE:

On the form of Ohm's law in magneto-hydrodynamics

PERIODICAL:

Prikladnaya matematika i mekhanika, v. 25, no. 4,

1961, 611-622

TEXT: A generalized form of Ohm's law is derived for a quasi-neutral medium consisting of electrons, ions and neutral atoms, in the presence of a space charge ρ_e . The limits of applicability of the obtained relationships are discussed, as well as the possible use (in concrete cases) of various forms of Ohm's law. Basic equations of the motion of the medium are then discussed, consisting of electrons, ions and neutral atoms. A very simple model is considered, in which the electrons, ions and neutral atoms are ideal gases. In which the electrons, ions and neutral atoms are ideal gases. The degree of ionization is defined as $\alpha = n / (n + n_{\lambda})$ (1.1) The degree of ionization is defined as $\alpha = n / (n + n_{\lambda})$ (1.1) and is the number of ions and n_{λ} that of neutral atoms. Formulas are given for the pressure ρ , the momentum J, velocity v, and for the motion of electron- and ion gases respectively. The equation of

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S/040/61/025/004/003/021 D274/D306

On the form of Ohm's law...

motion of the gas mixture is

 $m_{1}(n + n_{a}) \frac{dv}{dt} = - \text{grad } p - n'eE - \frac{e}{c} [nV_{e} + n' (v + v_{1})] \times H$ (1.8)

where m_i is the ion mass, n + n' is the number of electrons; formula $j = \sum_k n_k e_k v_k = -(n + n') e (v + v_i + v_e) +$

+ $ne(v + v_i) = - nev_e - n'e(v + v_i)$ was used which defines the current density, and $n^\prime \ll n$. The derivation of the generalized Ohm law is then looked at. A number of

assumptions are made $T \gg \max \{\tau, \tau_e, \tau_i\}$ $v_i \ll v_{ix}, |v_i + v_e| \ll v_{ex},$ $|v - v_a| < v_{ax}$

where T is the characteristic time, related to the characteristic dimension (length) L and the characteristic velocity U, by U = L/T; Vix, Vex, vax are the random velocities of ions, electrons, and neutral atoms, respectively; T, Te, Ti are the collision times between electrons and ions electrons and ions and atoms ween electrons and ions, electrons and atoms, and ions and atoms, respectively. The notations are introduced:

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On the form of Ohm's law...

$$j = - \text{nev}_e - \text{n'e(v + v}_i), j_i = \text{nev}_i$$

$$\mathcal{X} = \frac{1}{\omega_{e}\tau} = \frac{\mathrm{cm}_{e}}{\mathrm{eH}} \tau^{-1}, \quad \chi_{e} = \frac{1}{\omega_{e}\tau_{e}} = \frac{\mathrm{cm}_{e}}{\mathrm{eH}} \tau^{-1}, \quad \chi_{i} = \frac{1}{\omega_{i}\tau_{i}} = \frac{1}{2} \frac{\mathrm{cm}_{i}}{\mathrm{eH}} \tau_{i}^{-1}$$
(2.3)

where ω_e and ω_i are the Lermor frequencies of electrons and ions. Finally the generalized Ohm law is obtained: $- |\operatorname{grad} p_e + \beta (\alpha \operatorname{grad} p - \operatorname{grad} (p_i + p_e))| - ne(E + \frac{1}{c} v \times H) + \frac{1}{c} v \times H$

$$+\left[1-2\left(1-\alpha\right)\beta-\alpha\frac{\varkappa_{i}}{\varkappa_{e}+\varkappa_{i}}\frac{m_{e}}{m_{i}}\right]\frac{1}{\varepsilon}j\times H+\beta\alpha n'eE-\frac{1-\alpha}{\varepsilon}\left(\beta+\frac{\alpha}{1-\alpha}\frac{m_{e}}{m_{i}}\frac{\varkappa_{i}}{\varkappa_{e}+\varkappa_{i}}\right)en'v\times H+\left[\varkappa+\left(1-\beta\right)\varkappa_{e}-\frac{1-\alpha}{\varepsilon}\left(\beta+\frac{\alpha}{1-\alpha}\frac{m_{e}}{m_{i}}\varkappa_{i}\right)\frac{H}{\varepsilon}\left(j+n'ev\right)-\frac{1-\alpha}{H\left(\varkappa_{e}+\varkappa_{i}\right)}\left\{\left[\alpha\operatorname{grad}p-\frac{1-\alpha}{\varepsilon}\left(p_{e}+p_{i}\right)\right]\times H+\frac{1-\alpha}{\varepsilon}j\times H\times H-\left(1-\alpha\right)n'eE\times H\right\}=\frac{n'}{n}\frac{1-\alpha}{\varkappa_{e}+\varkappa_{i}}\left[\alpha\operatorname{grad}p-\operatorname{grad}\left(p_{e}+p_{i}\right)-\left(1-\alpha\right)n'eE+\frac{1-\alpha}{\varepsilon}\frac{m_{e}}{\varkappa_{i}+\varkappa_{i}}\right]\left\{\left[\alpha\operatorname{grad}p-\frac{m_{e}}{\varepsilon}\varkappa_{i}\right]+\frac{1-\alpha}{\varepsilon}j\times H\right\}$$

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3/040/61/025/004/003/021 D274/D306

On the form of Ohm's law...

(for partially-ionized gases). Simplifying assumptions are made which lead to simpler forms of the law. For not highly rarefied

For a completely ionized gas: $= \operatorname{grad} p_{e} - \operatorname{ne} \left(E + \frac{1}{c} v \times H \right) + \frac{1}{c} j \times H + \frac{2}{c} H \left(j - \rho_{e} v \right) = 0 \quad (2.13)$

In concrete problems, it may turn out that some of the terms in Eq. (2.12) are negligibly small, hence simpler forms of the generalized law can be used. The relative magnitude of the terms entering Eq. (2.12) is evaluated. It is found that the relative magnitude of these terms is determined by the value of the "ollowing dimensionless parameters:

parameters. $\omega_{e}\tau^{*}$, $\frac{1}{\alpha}\frac{\Omega}{\omega_{i}}$, $(\frac{1-\alpha)^{2}}{\chi_{i}}$, $\frac{2(1-\alpha)\gamma_{i}}{\alpha'i'}$, $\frac{1}{\alpha}\frac{\Omega}{\omega_{i}}(\omega_{e}\tau^{*})^{-1}$

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261²⁴ \$/040/61/025/004/003/021 D274/D306

On the form of Ohm's law ...

which are related to the physical properties of the medium and to the specific conditions of the problem; Ω is the characteristic frequency; Υ^* is defined by $\frac{1}{2\pi} = \frac{1}{2} + \frac{1}{2}$ (3.11)

The values of each of these parameters are discussed for various conditions of pressure, temperature, strength of magnetic field, etc; this leads to conclusions regarding the value of the various terms of Eq. (2.12) and their possible neglecting. Thus e.g. the parameter $[2(1-\alpha)/\alpha]\tau_i/T$ which determines the relative magnitude of the terms (σ/c) v × H and c⁻¹ j×H×H, can (by virtue of Eq.(2.1) be comparable to unity in case of a very small degree of ionization only; σ is the conductivity. It is noted that the above evaluation was carried out for quantities L, U and T which are the same for all mechanical and electromagnetic magnitudes; hence these estimates cannot be used for all problems, but other, analogous, estimates can be made. There are 9 references: 5 Soviet-bloc and 4 non-Soviet-bloc. The references to the English-language publications read as follows: T. Kihara, Macroscopic Foundation of Plasma Dynamics. J.

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26124

On the form of Ohm's law ...

S/040/61/025/004/003/021 D274/D306

Phys. Soc. of Japan, 1958, v. 13, no. 5; L. Steg and G.W. Sutton, The prospects of NHD power generation. Astronautics, 1960, v. 5, no. 8.

SUBMITTED:

April 7, 1961

Card 6/6

 LYUBIMOV, G. A

PHASE I BOOK EXPLOITATION

SOV/6191

Kulikovskiy, Andrey Gennadiyevich, and Grigoriy Aleksandrovich Lyubimov

Magnitnaya gidrodinamika (Magnetohydrodynamics) Moscow, Fizmatgiz, 1962. 246 p. 7500 copies printed.

Ed.: V. P. Korobeynikov; Tech. Ed.: K. F. Brudno.

PURPOSE: This book is intended for persons working in the field of magnetohydrodynamics.

COVERAGE: The book contains systematized basic principles of magnetohydrodynamics, presents relationships resulting from interaction of a conducting medium with an electromagnetic field, and investigates the possibility of obtaining exact solutions for magnetohydrodynamic equations. The author thanks M. N. Kogan and V. P. Korobeynikov for their advice. There are 134 references, about two-thirds of them Soviet.

Card 1/4/

S/040/62/026/003/015/020 0467/0561

26.2031

Lyubimov, G.A. (Moscow) AUTHOR:

On solving magnetohydrodynamic problems with anisotro-11111:

pic conductivity

Priklaunaya matematika i mekhanika, v. 26, no. 3, PERIODICAL:

1962, 530 - 541

TUXT: Proceeding from the generalized Ohm's law, the author obtains a single vector equation for the magnetic field, viz.

 $\underline{\wedge}$ H - $\alpha[(H \nabla) \text{rot H} - (\text{rot H} \nabla)H] = \frac{1}{2\pi} \cdot \frac{6H}{6t}$. (1.4)

This equation is analogous to the induction equation of magnetohydrodynamics. For stationary problems, Eq. (1.4) reduces to

 $\triangle H = \alpha[(H \nabla) \text{rot } H = (\text{rot } H \nabla)H] = 0.$

In view of applying the results to problems of flow in tubes and phannels under the effect of an external electromagnetic field, solutions to Eq. (2.1) are sought, which do not depend on the x-coordard (1/3)

S/040/62/026/003/015/020 D407/D301

On solving magnetohydrodynamic problems.. D407/D301

dinate. The boundary conditions are set up. Several particular solutions of the system of equations are considered. These solutions have the property that the density of the current and that of the electromagnetic force in the direction of the x-axis are proportional to the magnetic field-strength in that direction. As an application of these solutions, the flow of current from a flat electrone tion of these solutions, the influence of anisotropic conductivity on is discussed. Further, the influence of anisotropic conductivity on is flow in a rectangular channel, is considered. Two sides of the rectangle are electrodes, and the other two are dielectrics. The rectangle are electrodes, and the other two are dielectrics. The electric- and magnetic fields are crossed. In order to ascertain the electric- and magnetic fields are crossed. In order to ascertain the electric of anisotropic conductivity on the electromagnetic force, effect of anisotropic density of the electromagnetic force of anisotr

 $\tilde{J}_{X1} = -\frac{i \delta \tilde{U}}{\omega \tau} \frac{\delta H_{X}}{\delta z} = -I_{1} \omega \tau, \qquad (4.11)$

where j is the current density along the x-axis. In the second appro-Card 2/3

S/040/62/026/003/015/020

On solving magnetohydrodynamic problems.. D407/D301

ximation, one obtains a mixed problem for the Laplace equation. The dependence of I on the voltage, applied at the electrodes, is analyzed. Further, gas flow in a sylindrical channel is considered (also by the method of expansion in series). In the first approximation, the space-charge density differs from zero, whereas in the case of the rectangular channel, it was zero in the first approximation. As a more complicated example, flow in a rectangular channel is considered, with "point"-electrodes which are located at the points z=0 $y=\pm$ 0. In this case, too, the space charge differs from zero in the first approximation already. There are 4 figures.

ASSCCIATION: NII mekhaniki MGU (Scientific Research Institute of

Mechanics of Moscow State University)

SUBMITTED: January 25, 1962

Card 3/3

24.2120

s/040/62/026/004/012/013 D409/D301

26.1410 AUTHORS:

Kulikovskiy, A.G., and Lyubimov, G.A. (Noscow)

TITLE:

On magnetohydrodynamic shock-wave structure in a res

with anisotropic conductivity

PERIODICAL:

Prikladnaya matematika i mekhanika, v. 26, no. 4. 1962

791 - 792

TEXT: In the references (A.G. Kulikovskiy, O strukture udarnykh voln, PiM, this issue) it is shown that the width (thickness) of a shock-wave in a non-ideal medium may not vanish when all the discipation coefficients tend to zero. Below, such a shock wave is constructed. Ohm's generalized law is used in the following form:

$$cE + v \times E + \frac{c}{ne} \operatorname{grad} p_e = \frac{c}{c} \cdot i + \frac{c}{c} \frac{w\tau}{\tau} \cdot i \times U.$$

The equations for one-dimensional steady flow are set up. The matrix of the dissipation coefficients v_m^* and v_m^* is denoted by L_{ij} . One obtains for the width of the shock wave

On magnetohydrodynamic shock-wave ...

8/040/62/026/004/012/013 D403/D301

$$1 \sim v_{m}^{*} (1 + \pi^{2}) U^{-1},$$

where I is a characteristic velocity. This empression character if the lissipation coefficients some some, and, we have seen a wave behaves as follows

$$1 \rightarrow 0$$
, if $v_m^* \times^2 \rightarrow 0$; $1 \rightarrow 0$, if $v_m^* \times^2 \rightarrow 0$.

The latter case occurs only if $\omega\tau\to\infty$. With large $\omega\tau_*$ the colution of the shock-structure problem is periodic the which of each rich is of the order of $U\otimes\psi_m^{-1}$, and approaches some if the curve tion coefficients approach zero. A formula is given for the mass of increase of the entropy dP/dx. If $\psi_m^{-1}\to 0$, and $U_m^{-1}\otimes \dots$ remains finite, then $dP/dx\to 0$, and $1\to\infty$. Thereby, the solution approaches a periodic solution on any finite interval $\{x_1, x_2, x_3, x_4, x_5, x_6, x_7, x_8\}$ does not increase on this interval. Such a solution can be approached as a macroscopic analogue of the course of him solution approached plasma in the absence of discipation.

8/040/62/026/005/002/016 D234/D308 17 26 2231 Lyubimov, G. A. (Moscow) Formulation of the problem of magnetohydrodynamic AUTHOR: boundary layer Prikladnaya matematika i mekhanika, v. 26, no. 5, TITLE: TEXT: The author gives general considerations and relations which PERIODICAL: may be useful in formulating complicated problems. Only incompressible liquids are treated. It is found that one must include in the boundary layer equations the force (9) $f = \frac{\sigma}{c} (E + \frac{v}{c} x H) x H$ and the additional equations Card 1/2

Formulation of the ...

S/040/62/026/005/002/016 D234/D308

rot E = 0, div $E = 4\pi \rho_{\rm p}$

(2)

The boundary layer problem must then be generally solved together with the external problem. The author introduces some simplifications into Eq. (2) and formulates the boundary conditions for the external problem in such a way that the boundary layer protlem may be solved independently. Systems of equations for the boundary layer are deduced for the case when the magnetic Reynolds number $|R_{\rm mL}|$ is smaller or approximately equal to 1. The equations of the external problem are formulated separately for the cases $R_{\rm mL}\ll 1$ and $R_{\rm mL} \sim 1$. A simple example is given. The reference to N and N and N are formulated.

ASSOCIATION: MGU

SUBMITTED: June 22, 1962

Card 2/2

1,53,5

s/040/62/026/006/005/015 D234/D308

AUTHOR:

Lyubimov, G.A. (Moscow)

TITLE:

Magnetohydrodynamic boundary layer in a medium having anisotropic conductivity for small magnetic Reynold's

numbers

PERIODICAL: Prikladnaya matematika i mekhanika, v. 26, no. 6, 1962,

1077 - 1086

TEXT: General formulation of the problem is given and applied to the boundary layer on a semi-infinite plate, assuming that

$$mL = \frac{H_0^2 \sigma L}{c^2 \rho U (1 + \omega^2 \tau^2)} \leqslant 1$$
 (2.4)

and that the magnetic field is homogeneous and perpendicular to the plate. The final system of equations is

$$u_0 \frac{\partial u_1}{\partial x} + u_1 \frac{\partial u_0}{\partial x} + v_0 \frac{\partial u_1}{\partial y} + v_1 \frac{\partial u_0}{\partial y} - \frac{1}{R} \frac{\partial^2 u_1}{\partial y^2} = 1 - u_0$$
 (2.10)

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0

Magnetohydrodynamic boundary layer ... S/040/62/026/006/005/ct; D234/D308

$$\frac{\partial u_1}{\partial x} + \frac{\partial v_1}{\partial y} = 0, \quad u_0 \frac{\partial w_1}{\partial x} + v_0 \frac{\partial w_1}{\partial y} - \frac{1}{R} \frac{\partial^2 w_1}{\partial y^2} = \omega \tau u_0.$$

The last equation in it is reduced to an ordinary difference, the tion by putting $n = y \sqrt{(R/x)}$ and

$$w_1 = \omega \tau x \mathcal{T}(\eta)$$

Y was found by numerical integration. The deceleration of flow clonthe x axis decreases with increasing $\omega \tau$. The transverse velocity has a maximum at $\omega\tau=1$ and tends to 0 for $\omega\tau\to\infty$. The coefficients of longitudinal and transversal friction are computed up to terms of the first order in mL. There are 3 figures.

SUBMITTED: June 28, 1962

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L 15772-63 ACCESSION NR: EWT(1)/BDS/ES(W)-2

AFFTC/ASD/AFVIL/SSD Pab-4/Pi-4/16-4

AP3006126

8/0207/63/000/004/0078/0082

AUTHOR: Lyubimov, G. A. (Moscow)

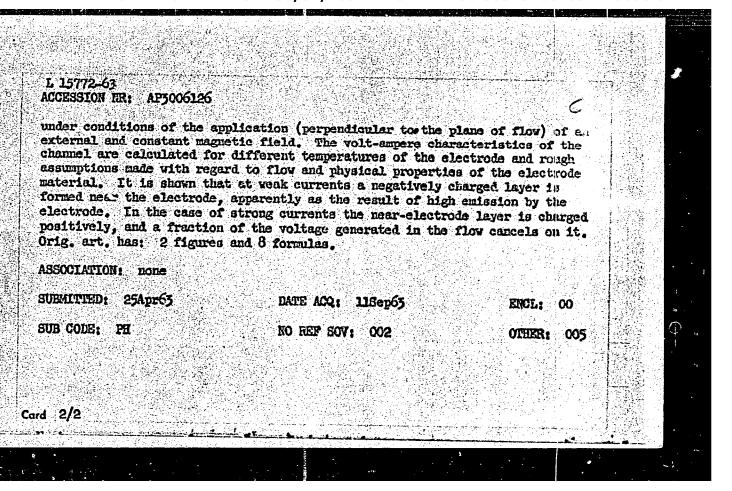
TITLE; Boundary conditions on contact surface of ionized gas-solid body

SOURCE: Zhurnal prikladnoy mekhaniki i tekhnicheskoy fiziki, no. 4, 1965, 78-82

TOPIC TAGS: interface boundary condition, contact potential difference, conductive gas flow, MED generator

ABSTRACT: It is shown that the boundary conditions for an electrode can be formulated when the physical properties of the electrode material are taken into account and that an exact formulation depends on the distribution of electrical values in the electrode layer, whose thickness is smaller than the length of the free path of the electron. The formulation is based on several assumptions with regard to the mechanism of electronic emission and the character of the nearelectrode layer. The correspondence of the accepted assumptions to the nature of the investigated problem is proved for each concrete case. For example, the problem of the flow of a conductive viscous gas in a plane channel is studied

Card



5/0207/63/000/005/00124/0034

ACCESSION NR: AP3014917

AUTHOR: Lyubimov, C. A. (Moscow)

TITLE: Electric potential changes near the walls of a channel during the motion of ionized gas in a magnetic field

SOURCE: Zhurnal prikl. mekhaniki i tekhn. fiziki, no. 5, 1963, 24-34

TOPIC TAGS: ionized gas electric potential, moving gas electric potential, electric potential near wall, magnetic field, ionized gas, ionized gas in magnetic field, moving ionized gas

ABSTRACT: The flow of a conducting viscous fluid in a plane channel is considered with constant velocity U and transverse, constant magnetic field H. For cold walls the conductivity is assumed to be a function of temperature across the channel and the magnitude of the wall potentials \bigcirc is derived for small currents, in a boundary layer with a straight line velocity profile, using Ohm's law and assuming $R_{\rm m} << 1$. The channel walls are then assumed to be hot electrodes, and the magnitude of each electron sheath is studied under an electric field E, including the

Card 1/2

Schottky effect on thermionic emission greater or less than the random electron plasma currents. Next, an external load of finite resistivity is applied, and expressions for the current densities at each electrode derived. The potential drop across the channel is then determined from

$$(R+r)j = \mathscr{E} - \varphi_{+}(j,T_{+}) + \varphi_{-}(j,T_{-}) + \Phi_{+} - \Phi_{-}$$

where Φ indicates work functions of each electrode and Φ , the sheath potentials. The first term on the right represents the induced field. This equation shows that in order to evaluate the current-voltage characteristics one should determine the sheath potentials in terms of the various plasma parameters. Orig. art. has: 29 formulas and 11 figures.

ASSOCIATION: none

SUBMITTED: 08Jun63

DATE ACQ: 27Nov63

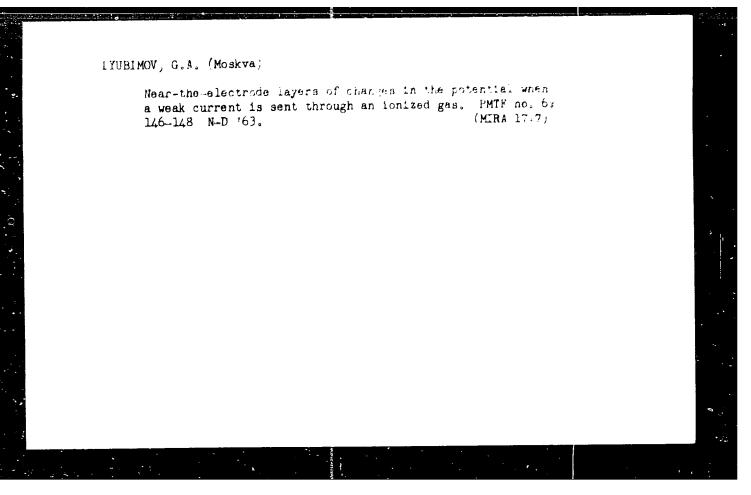
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NO REF SOV: 005

OTHER: 004

Card 2/2



S/0040/63/027/003/0509/0522

AUTHOR: Baranov, V. B. (Moscow); Lyubimov, G. A. (Moscow); Hu Yu-yin (Moscow)

TITLE: Calculation of the boundary layer on a dielectric plate in a flow of an incompressible, anisotropically conducting fluid in the presence of a homogeneous, transverse magnetic field

SOURCE: Prikladnaya matematika i mekhanika, v. 27, no. 3, 1963, 509-522

TOPIC TAGS: boundary layer, flow over flat plate, electrically conducting fluid flow, transverse magnetic field, flow in magnetic field, magneto-aero-dynamic effect

ABSTRACT: The results of the authors' previous works (Baranov, V. B. Prikl, mat, i mekh., v. 26, no. 6, 1962; Lyubimov, G. A. Prikl. mat, i mekh., v. 26, nos. 5 and 6, 1962) are applied to the solution of the boundary layer problem in weakly and fully ionized media. Under certain assumptions

Cord 1/3

the problem is reduced to the solution of a system of differential equations:

$$\ln \frac{\partial x}{\partial n} + \ln \frac{\partial x}{\partial n} - \frac{L}{3} \frac{\partial z}{\partial n} = -\frac{\partial x}{\partial p} + \text{wr}(-n + \text{min} + \text{min}^{x}) + \frac{1 + \text{min}^{2}}{\text{min}^{2}} \frac{\partial x}{\partial p^{9}}$$

$$u \frac{\partial v}{\partial x} + v \frac{\partial v}{\partial z} - \frac{1}{R} \frac{\partial^2 v}{\partial z^2} = -mL(\omega \tau u + v + E_x^0) - \frac{\omega \tau}{1 + \omega^2 \tau^2} \frac{\partial p_0}{\partial x}$$

$$\frac{\partial x}{\partial x} + \frac{\partial y}{\partial y} = 0.$$

Four different regimes of external flow are considered, and solutions are sought by linearization with respect to a certain parameter. The cyclotron frequency of ion rotation is assumed to be small in comparison to the ion collision frequency. The Thompson, Ettinghausen, and Leduc-Riggi effects are taken into account in the derivation of energy equations. Studies of the

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thermal boundary layer are presented, and graphs of the velocity and temperature profiles are given for various values of different parameters ω, τ, m*x, and Prandtl numbers from 0.1 to 0.01). Numerical calculations were made on the "Strela" computer at the MGU computing center. "The authors consider it their duty to thank M. N. Kogan and A. G. Kulikovskiy for their discussion of the results and useful critical comments and staff members G. S. Roslyakov and Ye. N. Starova of the MGU computing center for help in calculation." Orig. art. has: 12 figures and 49 formulas.

ASSOCIATION: none

SUBMITTED: 2LJan63 DATE ACQ: 23Jul63

ENCL: 00

SUB CODE: 00

NO REF SOV: 006

OTHER: 001

Card 3/3

BIMOV, G. A.	
"Electrical Potential Variation Layers Near Electrodes."	
report submitted for Intl Symp on Magnetonydrodynamics Electrical Power Generation, Paris, 6-11 Jul 64.	
Inst of Mechanics, Moscow State Univ.	

LYUBIMOV, G.A. (Moscow)

"Magnetohydrodymanic boundary layers".

report presented at the 2nd All-Union Congress on Theoretical and Applied Mechanics, Moscow, 29 Jan - 5 Feb 6h.

Card___

5/0179/64/000/001/0141/0142

AHTHOR: Baranov, V. B. (Moscow); Kulikovskiy, A. G. (Moscow); Lyubimov, G. A. (Moscow)

TITLE: The boundary layer on a flat plate in anisotropic magnetohydro-dynamics

SOURCE: AN SSSR. Izv. Otd. tekh. nauk. Mekhanika i mashinostroyeniye, no. 1, 1964, 141-142

TOPIC TAGS: flat plate, boundary layer, boundary layer condition, thermal boundary layer, Ettingshausen effect, aerodynamics

ABSTRACT: Expanding the subject of a previous report (Baranov, V. B., Izv, AN SSSR, OTN, Mekhanika I mashinostroyeniye, 1962, No. 6), the authors consider disturbances to an external flow caused by a boundary layer to show that temperature at the latter's boundary can be considered fixed despite the presence of the Ettingshausen effect. Further, it is shown that the inequality M \lesssim R (where M is Hartman's number, R is Reynold's number, as related to the characteristic length I along the plate) can be diminished and the form M \lesssim R can be used for the existence of the Blasius velocity profile. The thermal boundary layer is

ACCESSION NR: AP4018433

calculated with consideration of Ettingshausen's effect (see Fig. 1 in the Enclosure). "In conclusion, the authors express gratitude to M. N. Kogan for calling their attention to the problem and participating in evaluation of possible solutions". Orig. art. has: 1 figure and 10 formulae.

ASSOCIATION: none

SUBMITTED: 24Sep63

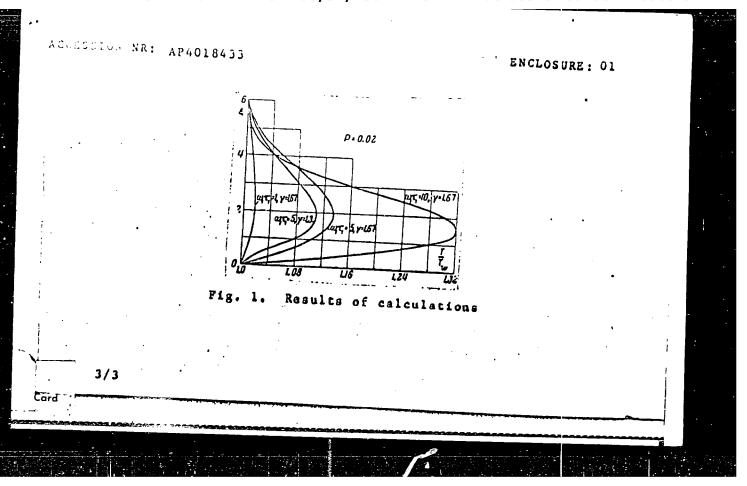
ATD PRESS: 3046

ENCL: 01

SUB CODE: ME

NO REF SOV: 003

OTHER: 000



S/0207/64/000/004/0010/0015

AUTHOR: Lyubimov, G. A. (Moscow)

TITLE: Some problems in theory of potential change through sheath

SOURCE: Zhurnal prikladnoy mekhaniki i tekhnicheskoy fiziki, no. 4, 1964, 10-15

TOPIC TAGS: plasma sheath, electron ion plasma, seeded plasma, thermionic emission, debye length

ABSTRACT: The author attempted to refine his previous theory on electric sheath formation (Prielektrodny*ye sloi immenentya potentsiala pri propouskanii slabogo toka cherez ionizovanny*y gaz. PMTF, 1963, No. 6) by including in his analysis such effects as collision ionization of atoms by electrons and acceleration of alectrons in sheath layers. The model consists of a thermionically enitting surface, a Debye length much smaller than the electron mean free path, and a linear potential drop E = P/d. The current balance equation between emitted electrons j_3 , plasma electrons j_6 , plasma ions j_1 , and collision ion.zed atoms j_4 is given by

 $/_{\circ} \exp\left\{\frac{4.39}{T} V \overline{E}\right\} + /_{i} + /_{i}^{\circ} - /_{\circ} \exp\left\{-\frac{e\varphi_{\bullet}}{kT}\right\} = /_{\circ}$

Cara 1/3

ADDIGGO : 146: AP4044714

For a potential change in the sheath greater than the ionization potential \mathbf{U}_k of the seed material added to the gas, the value of $\mathbf{j_i}$ * can be given by

$$I_{l}^{n} = eN \left\{ \frac{n_{R}Q_{k}}{n_{k}Q_{k} + n_{d}Q_{d}} C_{k} \left(\varphi_{+} - U_{k} \right) + \frac{n_{d}Q_{d}}{n_{k}Q_{k} + n_{d}Q_{d}} C_{d} \left(\varphi_{+} - U_{d} \right) \right\}$$

$$C_{k} = 0 \quad \text{at} \quad \varphi_{+} < U_{k}, \qquad C_{d} = 0 \quad \text{at} \quad \varphi_{+} < U_{d}$$

Q- electron-atom collision cross section. To this is added the current due to secondary electron emission by ions bombarding the cathode surface, given by $\gamma = j_e^{0}/j_i^{0}$, leading to the final expression for j given by

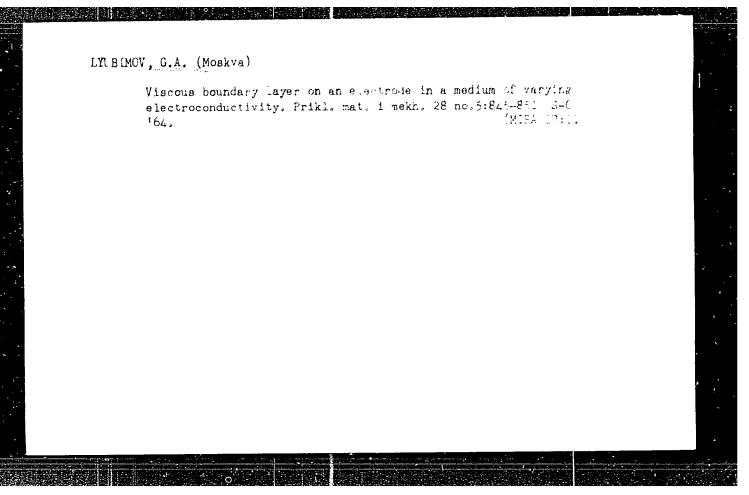
$$i = j_{s} \exp\left\{\frac{4.39}{T} \sqrt{E}\right\} + (j_{t} + j_{t} + j_{t} + (1 + \gamma_{d}) j_{t} + (1 + \gamma$$

Comparison of above theory with experimental data collected in argon at T=2200K and one atm pressure with additions of potassium seed indicated a considerably better agreement with experimental data than for analysis done without considering collisional ionization. Orig. art. has: 16 formulas and 1 figure.

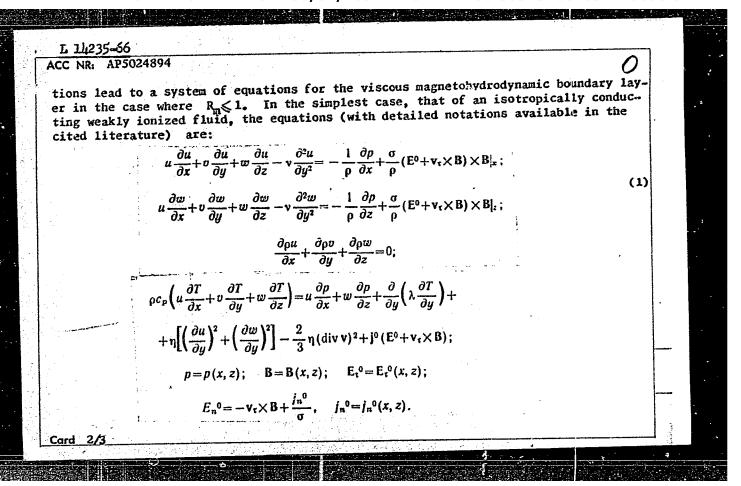
ASSOCIATION: none

Card 2/3

ACCESSION NR: AP4044714
SUBMITTED: 19Apr64
SUB CODE: ME,GP
NO REF SOV: 005
OTHER: 003



ACC NR: AP5024894 UR/0382/65/0	
AUTHOR: Lyubimov, G.A.	
ORG: None	В
TITLE: Magnetohydrodynamic boundary layer	
SOURCE: Magnitnaya gidrodinamika, no. 3, 1965, 3-20	
TOPIC TAGS: magnetohydrodynamic theory, magnetohydrodynamic bou	ndary layer
ABSTRACT: A review of selected published work on the theory of hydrodynamic boundary layer is presented for the cases where the number, $R_{\rm m}$, is less than unity, and the Reynold's number, R , i unity: $F\gg 1 \gg R_{\rm m}$. These cases are of interest in many theoret problems. Attention is directed to the mathematical and physical problems, with emphasis upon qualitative considerations. A list serves as a source of quantitative data and as a reference board tative discussions of the author. Consideration starts with a reand mathematical consequences of imposing suitable limiting conditions parameters. Estimates of the boundary layer thickness and of magacross the boundary layer lead to certain clarifications; for intended to conditions for external problem dominance via its influence of the boundary layer (internal problem) equations. These	s much larger than ical and experimental if fundamentals of the of some 54 references for several quantitions on the physical litions on the problem gnetic field changes stance, to a recognition on magnetic



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The arbitrary functions in the equations (1) of the internal problem (p; B; E and j), are determined by the solution of the external problem. Equations for an environment with an anisotropy of electrical conductivity, and equations for the case of strong ionization are available in the cited literature. The imposition of further parameter restrictions (e.g.: E= 0, or B = const. everywhere) - lead to additional decisive simplification of the external problem. The corresponding internal problem, defined by the boundary layer equations, can then be considered as an efficient point of departure for a review of basic boundary layer problem solutions. The author's review starts from this basic point and includes many papers. Among his comments are those on the discovery of a decrease in friction and in heat exchange with an increase in the magnetic field; and on the appearance of normonotonous velocity profiles. Research on the anisotropic and highly ionized cases is also reviewed. The author concludes that the existing studies of the magnetohydrodynamic viscous boundary layer refer, as a rule, to specific segregated problems and are aimed at quick qualitative answers. It is thus clear that the systematic investigation of the overall problem belongs still to the future. For instance, an approach to the magnetohydrodynamics of the boundary layer in the case of highly ionized environment and absence of plasma thermal equilibrium still needs the construction of a basic system of equations. Considerations for the advancement of magnetohydrodynamic boundary layer research are contributed. Orig. art. has 15 figs., 12 formulas.

SUB CODE: 20

SUBM DATE: 26Nov64/

ORIG REF: 024

OTH REF: 030

3B Card 3/3

L 8485-66 EWT(1)/EWT(m)/ETC/EPF(n)-2/T IJP(c) ΑT ACC NR: AP5021903 SOURCE CODE: UR/0207/65/000/004/0045/0053 AUTHOR: Lyubimov, G. A. (Moscow) DRG: none TITLE: Near-<u>electrode</u> layers in "hot" electrodes SOURCE: Zhurnal prikladnoy mekhaniki i tekhnicheskoy fiziki, no. 4, 1965, 45-53 21, 44, 55 TOPIC TAGS: ionized gas, electrode, boundary layer plasma ABSTRACT: A semi-empirical theory of the properties of ionized gas near hot electrode surfaces is constructed. Several models are considered for which current density, particle velocity distribution and the Poisson equation yield the electric field near the electrodes, boundary layer thickness and current densities. The appropriate conditions are discussed under the assumption of equilibrium and the voltage-current curves are plotted. A variety of cases arises from the multitude of possible MHD concepts to which the developed theory is applied. It is shown that a choice of empirical constants is somewhat ambiguous; this leads to more than one solution, all within the experimental data. Other examples of comparison between the derived and measured voltage-current curves are given. Only a tentative comparison is made since not all parameters required by the theory were measured in all instances. Orig. art. has: 5 figures, 21 formulas. SUB CODE: 20/ SUBH DATE: 20Jun65/ ORIG REF: 005/ OTH REF: COS

L 16703-66 EWP(m)/EWT(1)/EWA(1) WW ACC NR: AP5003209 SOURCE CODE: UR/0382/65/000/004/0080/0084

AUTHOR: Lyubimov, G. A.; Semenova, I. P.

85. 83.

ORG: none

R.

TITLE: Formulation of the problem of a viscose boundary layer on a cold electrode

SOURCE: Magnitnaya gidrodinamika, no. 4, 1965, 80-84

TOPIC TAGS: temperature dependence, Maxwell equation, current density, electrode, conducting gas, boundary layer flow, electric field, fluid flow, electric conductivity, conductive fluid 21,44,55 ABSTRACT: The effect of the axial electric field in the core of a fluid stream

ABSTRACT: The <u>effect of the axial electric field</u> in the core of a fluid stream characterized by temperature-dependent electrical conductivity is investigated. In contrast to the results of other authors, it was found that cold electrodes in contact with the conducting gas cause a significant decrease of conductivity in the outermost regions. This phenomenon leads to the development of potential differences with amplitudes comparable to those of the applied field across the stream. To account for this effect, Maxwell's and Ohm's equations are solved for the associated current densities. It is shown that the resulting currents induce Joule heating

UDC: 533.95 : 538.4

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APPROVED FOR RELEASE: 08/31/2001

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which affects the fl ratios of the resist	ow. The characte	ristics of th	e flow are	plotted for	
ratios of the resist transfer and friction	ance of boundary	layers to tha	it of the in	mer lavers.	Heat
transfer and friction cient to estimate the	n are not consider	red, since th	e conventio	nal approach	neat h ia cu <i>eet</i>
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L 07017-67 EWT(1)/EWP(m)/EWT(m) IJP(c) DJ	
ACC NR: AP7001061 SOURCE CODE: UR/0421/66/000/003/0003	0/0011
AUTHOR: Lyubimov, G. A. (Moscow	56
ORG: none	B
TITLE: Averaging of magnetohydrodynamic flows and the applicability of a hydrau approximation for the calculation of magnetohydrodynamic flows through channels SOURCE: AN SSSR. Izvestiya. Mekhanika zhidkosti i gaza, no. 3, 1966, 3-11	lic
TOPIC TAGS: magnetohydrodynamics, nonuniform flow	
ABSTRACT: As engineers are beginning to use the methods of hydraulic calculations for the design of definite magnetohydrodynamic devices (using the analysis of overall characteristics like the efficiency, power at the external load, etc., supplied by hydraulic calculations), it is becoming clear during investigations in gas dynamics of the same nonuniform flow that the results of hydraulic calculation may differ by as much as 10, depending on the method of averaging. Consequently, the author carries out a comprehensive investigation of the problem of averaging of magnetodynamic flows and shows, for selected examples,	
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that the nonuniformity of magnetohydrodynamic flows is much stronger than in gas dynamic currents because of the nonuniformity of the force and thermal interactions of currents within the flows. These increased nonuniformities lead, in turn, to even larger fluctuations in the results of hydraulic calculations, and the author concludes that there is a definite need for the development of hydraulic calculational methods based on nonuniform (in the hydrodynamic and electrodynamic sense) canonical flows. Such an approach appears in many cases as the only one applicable for practical design of magnetohydrodynamic equipment. Orig. art. has:

3 figures and 2 formulas. [JFRS: 37.75]

SUB CODE: 20 / SUHM DATE: 10Feb65 / ORIG REF: 006 / OTH REF: 002

124-57-1-1210

Translation from: Referativnyy zhurnal, Mekhanika, 1957, Nr., p.lot (USSR)

AUTHORS: Pochtovik, G. Ya., Lyubimov, G. D., Likverman, A. I.

TITLE: Investigation of the Strength, Rigidity, and Crack Resistance of

Reinforced-concrete Structures Based on "Keramzit" Clayey Filler Gravel (Issledovaniye prochnosti, zhestkosti i treshchinoustoychivosti zhelezobetonnykh konstruktsiy na keramzitovom

gravii)

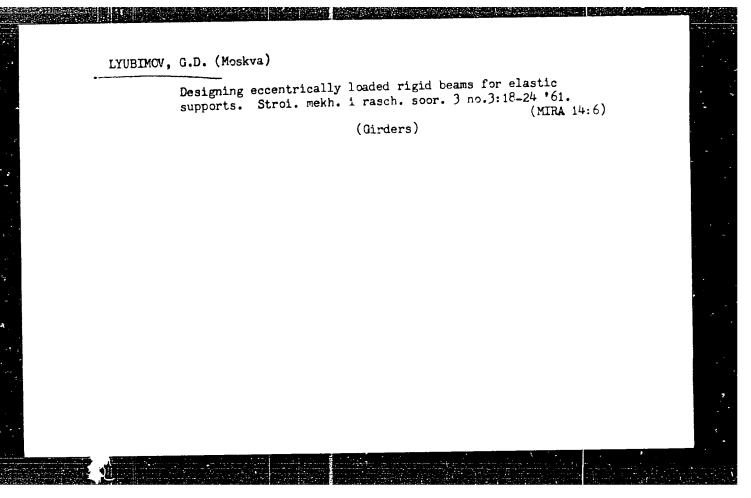
PERIODICAL: Tr. Mosk. avtomob.-dor. in-ta, 1956, Nr 18, pp 231-240

ABSTRACT: Bibliographic entry

1. Box beams--Stresses--Test results 2. Box beams--Vibration.

--Mathematical analysis

Card 1/1



LYUBIMOV, GP

56-6-12/47

AUTHORS:

Lyubimov, G. P., Khokhlov, R. V.

TITLE:

On the Polarization of a Mclecular Beam by an Alternating Field With Changing Amplitude and Phase (O polyarizatsii molekulyarnogo puchka peremennym polem s izmenyayushchimisya amplitutoy i fazoy)

PERIODICAL:

Zhurnal Eksperimental'noy i Peoreticheskoy Fizki, 107, Vol. 35, Nr 6, pp. 1396 - 1402 (USSR)

ABSTRACT:

The principle upon which the operation of a molecular generator is based consists in the fact that the molecule flying through the resonator enters into interaction with its electric field. On this occasion the excited molecule will pass its excess energy on to the resonator. For the analysis of all possible, stable, steady oscillations it is necessary to determine the polarization of the beam if this beam is subjected to the action of an alternating field with changing amplitude and phase.

The paper contains the following chapters:

1.) Derivation of the initial equation

2.) Discussion of the case in which the phase of the loting field is constant

Card 1/2

3.) Discussion of the case in which the amplitude and frequency of